

AD0637504

AFFDL-TR-66-52



AFFDL-TR-66-52

## FUNDAMENTALS OF STATISTICAL ENERGY ANALYSIS OF VIBRATING SYSTEMS

TECHNICAL REPORT No. AFFDL-TR-66-52

MAY 1966

AIR FORCE FLIGHT DYNAMICS LABORATORY  
RESEARCH AND TECHNOLOGY DIVISION  
AIR FORCE SYSTEMS COMMAND  
WRIGHT-PATTERSON AIR FORCE BASE, OHIO

20080815 138

Distribution of this document is unlimited.

(Prepared under Contract No. AF 33(615)-2649 by  
Bolt Beranek and Newman, Inc., Cambridge, Massachusetts;  
Eric E. Ungar, Author)

## NOTICES

When Government drawings, specifications, or other data are used for any purpose other than in connection with a definitely related Government procurement operation, the United States Government thereby incurs no responsibility nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data, is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use, or sell any patented invention that may in any way be related thereto.

Copies of this report should not be returned to the Research and Technology Division unless return is required by security considerations, contractual obligations, or notice on a specific document.

# **FUNDAMENTALS OF STATISTICAL ENERGY ANALYSIS OF VIBRATING SYSTEMS**

*ERIC E. UNGAR*

AD637504

**Distribution of this document is unlimited.**

## FOREWORD

This report was prepared by Bolt Beranek and Newman Inc., Cambridge, Massachusetts, for the Aerospace Dynamics Branch, Vehicle Dynamics Division, AF Flight Dynamics Laboratory, Wright-Patterson Air Force Base, Ohio 45433, under Contract No. AF 33(615)-2649. The research performed is part of a continuing effort to provide advanced techniques in the application of random process theory and statistics to vibration problems. This effort is part of the Research and Technology Division, Air Force Systems Command's exploratory development program. The contract was initiated under Project No. 1370, "Dynamic Problems in Flight Vehicles," Task No. 137005, "Prediction and Control of Structural Vibration." Mr. R. G. Merkle of the Vehicle Dynamics Division, FDDS was the project engineer.

This report covers work conducted from July 1965 to February 1966. Contractor's report number is 1350. Manuscript released by the author in February 1966 for publication as an AFFDL Technical Report.

The author is indebted to his colleagues, Drs. Preston W. Smith, Jr., Richard H. Lyon, Jerome E. Manning, and Terry D. Scharton, for the ideas they contributed to the subject matter of this report and for their many helpful suggestions during its preparation.

This report has been reviewed and is approved.

*Walter J. Mykytow*  
Walter J. Mykytow  
Asst. for Research and Technology  
Vehicle Dynamics Division



## ABSTRACT

The concepts and relations which form the theoretical foundation of the statistical energy approach to vibration analysis are delineated. The utility of this approach for dealing with complex systems is discussed, and its range of applicability is indicated.

The important properties of the modes of vibrating systems are reviewed and used to exhibit relations between modal responses and average responses of total systems. It is demonstrated that under some conditions, which are often approximated in actual systems, the average rate of flow of energy from one mode to another is proportional to the difference in the modal energies. It is also shown that under some conditions the average rate of flow of energy between two sets of modes (representing groups of modes of two coupled systems in a given frequency band) is proportional to the difference in the set-average modal energies. Examples are presented which indicate how these relations permit one to obtain approximate solutions to complex problems simply, on the basis of energy conservation considerations.

Available extensions of the major concepts developed in detail are mentioned, and references to the current literature are given.

## TABLE OF CONTENTS

|  | <u>Page</u> |
|--|-------------|
| INTRODUCTION .....   | 1           |
| MODES OF CONTINUOUS SYSTEMS .....                                      | 4           |
| Mode Shapes and Natural Frequencies .....                              | 5           |
| Orthogonality and Normalization of Mode Shapes...                      | 6           |
| Representation of Deformations in Terms of Modes.                      | 8           |
| Modal Analysis of System Motions .....                                 | 9           |
| Spatial Averages of Deformations; System Kinetic<br>Energy .....       | 10          |
| Summary and Conclusions .....  | 12          |
| POWER FLOW BETWEEN TWO MODES .....                                     | 13          |
| Admittances and Impedances .....                                       | 14          |
| Implications of Conservative Coupling .....                            | 15          |
| Time-Average Modal Kinetic Energies .....                              | 17          |
| Calculation of Net Power Flow by Superposition<br>of Flows .....       | 21          |
| General Expression for Power Flow .....                                | 23          |
| Relation Between Power Flow and Kinetic Energies.                      | 24          |
| Power Flow for Weak Coupling .....                                     | 25          |
| Power Flow for Stiffness and Gyroscopic<br>Coupling .....              | 26          |
| Finite Frequency Band Excitation .....                                 | 27          |
| Summary and Conclusions .....  | 28          |
| POWER FLOW BETWEEN TWO SETS OF MODES .....                             | 30          |
| Power Flow Between Mode Sets in Terms of<br>Set-Average Energies ..... | 31          |
| Modal Energy-Independent Coupling Factors .....                        | 33          |
| Uniform Coupling .....   | 33          |
| Uniform Modal Energies .....   | 34          |
| Power Flow in the Ensemble Average .....                               | 37          |
| Summary and Conclusions .....  | 39          |

| TABLE OF CONTENTS (continued)   | <u>Page</u> |
|---|-------------|
| LOSS FACTORS, RESISTANCES, AND MODAL DENSITIES .....                    | 41          |
| Loss Factors .....  | 41          |
| Coupling Loss Factors .....   | 43          |
| Loss and Coupling Resistances .....                                     | 44          |
| Modal Densities .....   | 46          |
| SOME SPECIAL RESULTS FOR TWO COUPLED MODE SETS .....                    | 51          |
| Vibrations of Indirectly Excited Mode Sets .....                        | 51          |
| Effect of Coupling on Vibrations of Directly<br>Excited Mode Sets ..... | 53          |
| Interaction of Sound and Structures .....                               | 54          |
| CONCLUDING REMARKS .....  | 58          |
| REFERENCES .....  | 60          |

| ILLUSTRATIONS AND TABLES                              | <u>Page</u> |
|---|-------------|
| Figure 1 Two Coupled Modes .....                      | 63          |
| Figure 2 Power Flow Diagram for Two Coupled Modes ... | 63          |
| Figure 3 Two Coupled Sets of Modes .....              | 64          |
| Figure 4 Power Flow Between Two Sets of Modes.....    | 65          |
| Table 1 Modal Densities of Some Uniform Systems ....  | 48          |



## INTRODUCTION

Recent years have seen a continuation and acceleration of the trend toward higher performance vehicles, toward increased propulsion system power, and toward more highly sophisticated equipment and instrumentation. This trend has brought with it a greater incidence of problems associated with high frequency vibrations and has caused aerospace engineers to concern themselves increasingly with vibrations at frequencies considerably above the fundamental structural resonances.

Classically, vibration engineers have focused their attention on low-frequency oscillations, since the lowest few vibration modes are generally the ones which are associated with the greatest deflections, the highest stresses, and gross structural failures. However, the rather complete arsenal of analysis techniques which have been developed for dealing with low-frequency vibration problems contains none which can deal simply and effectively with most high-frequency problems, such as those of importance in relation to sonically induced fatigue, instrumentation performance, or noise transmission.

Although the classical methods are valid in principle at all frequencies, their use is in fact very often impractical for high frequencies, particularly for randomly excited complex structures. The classical approach consists of determining the natural modes, of calculating the responses of these modes to a specified excitation of interest, and of superposing these responses to determine the total structural response. Continuous structures have an infinite number of modes, but generally only the lowest few of these are of importance in low frequency vibrations, so that in these cases one needs to consider only those few modes. At high frequencies, however, a frequency band of interest usually encompasses the resonances of a large number of modes, and one must consider the responses of all of these modes in calculating the structural vibrations in that band.

Such classical multimodal analyses may require an amount of computation which is so great that it exceeds the capacities of some of today's largest computers. But, one also encounters another, perhaps more significant, limitation on the utility of such analyses. Since the shapes of the higher modes are much more sensitive to details of the structure than are those of the lower modes, one must be able to describe the structural and material properties



with greater precision in order to be able to perform higher mode calculations meaningfully. The required precision can generally be achieved only for fictitious structures, in view of the manufacturing tolerances and the variation in material properties one inevitably encounters in realistic cases.

If one could perform a high-frequency multimodal analysis meaningfully, one would still be faced with the problem of interpreting the large mass of results that is generated in such a set of calculations. These results would be descriptions of the motions, within the frequency band of interest, of a large number (ideally all) of the points of the structure. In general, these motions will be complex and different for different points, and one will need to perform additional calculations, e.g. to determine average values of various response properties, in order to reduce the large amount of information to tractable size.

The statistical energy analysis approach to structural vibrations problems, pioneered by R. H. Lyon, P. W. Smith, Jr., and I. Dyer, was developed in response to the need for a simple means for understanding and estimating significant properties of multimodal vibrations of complex systems. This approach was spawned by the realization that averaging initially, and then carrying out calculations in terms of average quantities, should lead to results much more readily than the classical approach, which involves much initial detailed calculation and subsequent averaging.

The statistical energy approach to some extent is analogous to the "room acoustics" approach for dealing with sound in architectural spaces. In the latter approach one does not attempt to solve the acoustic wave equation in detail for the complex spaces involved; rather, one analyzes the average behavior of energy variables and interprets the results in terms of averages of dynamic variables (e.g., pressures, velocities). In the statistical energy approach one similarly does not concern oneself with a detailed solution of the system equations of motion, but one analyzes average energy distributions and relates those to average response variables.

Although the term "statistical energy method" has come into somewhat popular use at this writing, there really exists no "method" consisting of a well-defined technique leading to a guaranteed result (like the Rayleigh-Ritz or the Holzer methods, for example). There does exist an



approach, which - like room acoustics - is one of a set of tools, the intelligent use of which can give one approximate, but simple and useful answers to problems which could not otherwise be solved realistically.

The fundamentals of the statistical energy analysis approach are scattered in the technical literature, generally are couched in language which may discourage the non-specialist, and in some cases have been only implied, and not spelled out in detail. Consequently, this potentially highly useful approach has not come to be used as widely as one may have expected on the basis of its simplicity of application. It is the purpose of this report to fill the apparent need for a presentation of the fundamentals of the statistical energy analysis approach in easily accessible form, for an explanation of its area of applicability, and for an illustration of how these fundamentals may be used.

The first of the following sections reviews the concepts and properties of modes of structural vibrations, and points out some useful relations between modal response and total average response properties. The second section derives the basic relations that govern the exchange of energy between two coupled modes. The third section generalizes this relation to enable one to determine the flow of energy from one set of modes, representing one structure or fluid system, to another set of modes, representing another such system. The fourth section relates some of the parameters introduced in the modal analyses to more commonly employed, and more easily estimated and measured system parameters. The fifth and final section illustrates the application of the statistical energy approach to a special class of problems.

The reader who does not have the inclination to wade through the considerable amount of mathematical manipulation which appears in this report should not be discouraged. He should ignore all those details and concentrate on the results and conclusions indicated in the "Summary and Conclusions" subsections. The strong point of the statistical energy approach, after all, is the simplicity with which the central results can be stated and applied.

## MODES OF CONTINUOUS SYSTEMS

If an undamped linear elastic system vibrates in the absence of external forces in such a way that all points in the system move sinusoidally in time at the same frequency, then the system is said to vibrate in a "natural mode" - or "mode", for short. This type of vibration is a very special member of the class of all possible types of free vibrations, and possesses some useful properties, several of which are reviewed in the present section.

The various points in an undamped system vibrating in a natural mode generally oscillate at different amplitudes, but are either completely in phase or out of phase with each other. All points reach their maximum excursion (some in the negative and some in the positive direction) at the same instant, all points pass through their equilibrium positions simultaneously, and all points reach any fraction of their maximum excursions simultaneously. Thus, a natural mode oscillation may be characterized by a spatial distribution of deformations (e.g., by the distribution of deflections at any suitable instant), multiplied by a sinusoid in time and a constant which is indicative of the amplitudes.

The spatial distribution of deformations is called a "mode shape". Functions describing a given mode shape may differ from each other by a multiplicative constant, since the definition of the mode shape is arbitrary to that degree. However, once a mode shape function has been selected, the constant by which one must multiply this function so that it represents a given modal vibration is fully determined. This latter constant is called "modal amplitude"; a specified modal amplitude and mode shape function together define a unique distribution of deformations.

The frequency at which a system in absence of external forces vibrates with a given mode shape is called the "natural frequency" corresponding to that mode. A continuous elastic system can vibrate in an infinite number of different modes, with an infinite number of corresponding natural frequencies.

The discussion of modes presented here is of necessity limited in scope and rigor. A clearly written more complete treatment may be found in Reference 1; a collection of information on the modal properties of simple systems is available in Reference 2.



## Mode Shapes and Natural Frequencies

The linear partial differential equations that govern the motions of undamped continuous elastic systems (such as strings, membranes, shafts, beams and plates) in absence of external forces may be written in the form

$$\mu \ddot{u} + \tilde{K} u = 0 \quad . \quad (1)$$

Here  $\tilde{K}$  is a linear differential operator which involves only the spatial coordinates and represents the stiffness characteristics of the system, and  $\mu$  is a function of only the coordinates and describes the mass distribution of the system. The symbol  $u$  denotes the deflection of the system from equilibrium as a function of time  $t$  and of the spatial coordinates, and the dot atop a symbol indicates differentiation with respect to time.

[For example, for a shaft in torsion  $u=u(x,t)$  denotes the torsional angular deflection as a function of the position along the shaft and of time. The corresponding operators for Eq. (1) are

$$\mu = \rho J, \quad \tilde{K} = - \frac{\partial}{\partial x} \left( K_s G \frac{\partial}{\partial x} \right) \quad ,$$

where  $\rho$  denotes the material density,  $J$  the polar moment of inertia of the cross-sectional area,  $K_s$  the torsional constant of the crosssection,  $G$  the shear modulus of the material. In the most general case all of these parameters may be functions of the coordinate  $x$  measured along the length of the shaft.

Similarly, for a flat plate in flexure,  $u=u(x,y,t)$  denotes the displacement of the plate normal to its midsurface, as a function of coordinates  $x,y$  along that midsurface and of time. Here

$$\mu = \rho_s(x,y), \quad \tilde{K} = \nabla^2 (D \nabla^2) \quad ,$$

where  $\rho_s$  denotes plate mass per unit surface area, and  $D$  represents the plate flexural rigidity, usually given as  $Eh^3/12(1-\nu^2)$  for homogeneous plates of thickness  $h$ , Young's modulus  $E$  and Poisson's ratio  $\nu$ . The symbol  $\nabla^2$  represents the "Laplacian" operator; in the usual Cartesian coordinates  $\nabla^2 = \frac{\partial^2}{\partial x^2} + \frac{\partial^2}{\partial y^2}$ . In the most general isotropic case,  $\rho_s$  and  $D$  may both vary with  $x$  and  $y$ .]



Natural modal vibrations are defined as free (unforced) vibrations in which all points move in unison, sinusoidally in time, at the same frequency. Such vibrations correspond to solutions of Eq. (1) which are of the form<sup>1</sup>

$$u(x,y,t) = \psi(x,y) [B \cos(\omega t + C)] \quad , \quad (2)$$

in which the spatial and time dependences are separated and where B and C are constants. Substitution of Eq. (2) into (1) yields the time-independent partial differential equation

$$\tilde{K}\psi - \omega^2 \mu \psi = 0 \quad (3)$$

which the function  $\psi(x,y)$  of the spatial coordinates must satisfy.

One finds that solutions  $\psi$  of Eq. (3), which satisfy the boundary (edge) conditions applicable in a given case and which are not identically zero, exist only for certain values of the frequency  $\omega$ . These values are called the natural frequencies, and the corresponding solution functions  $\psi(x,y)$  are called the mode shapes of the system. To each n-dimensional system ( $n=1,2,3$ ) there correspond n-fold infinite sets of natural frequencies and mode shapes. To each two-dimensional system (plate, membrane) there correspond doubly infinite sets of natural frequencies  $\omega_{jk}$  and mode shapes  $\psi_{jk}(x,y)$ ;  $j,k = 1,2,\dots$ , which satisfy Eq. (3) or

$$\tilde{K}\psi_{jk}(x,y) - \omega_{jk}^2 \mu \psi_{jk}(x,y) = 0 \quad . \quad (4)$$

### Orthogonality and Normalization of Mode Shapes

The mode shapes of systems whose edges are elastically constrained against lateral and rotational deformation<sup>2</sup> possess

<sup>1</sup>From here on the notation used is that for two-dimensional systems. Expressions applicable to one- and three-dimensional systems can be deduced by direct analogy. (Also, see Ref. 2.)

<sup>2</sup>Zero and infinite values of constraint stiffness are permitted, but in the general case the boundaries exert restoring forces which are proportional to the deformations and/or restoring moments which are proportional to the rotations. Infinite lateral stiffness and zero rotational boundary stiffness corresponds to the usual "simple support" conditions. The case where both stiffnesses are infinite corresponds to clamped edges; that where both are zero corresponds to free edges. (p. 269, Ref. 1.)

the useful property of "orthogonality". Mathematically, this means that the mode shapes satisfy

$$\iint_{A_s} \mu(x,y) \psi_{jk}(x,y) \psi_{j'k'}(x,y) dA_s = \begin{cases} m\psi_{jk} & \text{for } j=j', k=k' \\ 0 & \text{otherwise} \end{cases} \quad (5)$$

where the integration is carried out over the entire elastic system (here shown as over the entire area  $A_s$  of a two-dimensional system). In other words, if the integral of Eq. (5) involves two different modes, then the result is zero, but if it involves the same mode twice, then the result is a finite number. This number has been denoted by  $(m\psi_{jk})$ , where  $m$  denotes the total mass of the system, and satisfies

$$m\psi_{jk} \equiv \iint_{A_s} \mu(x,y) \psi_{jk}^2(x,y) dx dy, \quad (6)$$

in accordance with Eq. (5).

One may observe that multiplication of  $\psi$  by an arbitrary constant does not affect Eq. (3) and that hence this equation defines  $\psi(x,y)$  only within an arbitrary multiplicative constant. Equation (6) therefore involves a similar arbitrariness. It is generally useful for analysis purposes to "normalize" the mode shapes  $\psi_{jk}(x,y)$ , - that is, to set down some more or less arbitrary rule for determining the multiplicative constant to be used in a given analysis. One commonly used rule (but by no means the only one) requires  $\psi_{jk}=1$  for all  $j$  and  $k$ , so that  $m\psi_{jk}$  in Eq. (6) reduces to the total system mass  $m$ . If this rule is used, then the mode shape functions satisfy

$$\iint_{A_s} \mu(x,y) \psi_{kj}^2(x,y) dx dy = m = \iint_{A_s} \mu(x,y) dx dy \quad (7)$$

in addition to Eq. (4).



[For example, for torsional vibrations of a uniform shaft clamped at one end ( $x=0$ ) and free at the other ( $x=L$ ), one finds

$$\psi_j(x) = \sqrt{2} \sin[(2j-1)\pi x/2L], \quad \omega_j = [(2j-1)\pi/2L]\sqrt{K_s G/\rho J}.$$

For a uniform rectangular plate extending from  $x=0$  to  $x=a$  and from  $y=0$  to  $y=b$ , and simply supported on all four edges one finds

$$\psi_{jk}(x,y) = 2 \sin(j\pi x/a) \sin(k\pi y/b) \quad ,$$

$$\omega_{jk} = \pi^2(j^2/a^2 + k^2/b^2) \sqrt{D/\rho_s h} \quad .]$$

### Representation of Deformations in Terms of Modes

Any mathematically "well-behaved" system deflection  $u(x,y,t)$  may be expressed as a sum

$$u(x,y,t) = \sum_{j=1}^{\infty} \sum_{k=1}^{\infty} U_{jk}(t) \psi_{jk}(x,y) \quad (8)$$

involving the mode shapes  $\psi_{jk}$  and "modal amplitude coefficients"  $U_{jk}(t)$ , which depend only on time (Refs. 1,3). To evaluate these coefficients one may proceed in a manner which is analogous to that used to obtain a Fourier series expansion of a given function. If one multiplies both sides of Eq. (8) by  $\mu(x,y)\psi_{j',k'}(x,y)$  and integrates over the entire elastic system, then one finds in view of the orthogonality relation (5) that only one of the integrals of the sum will not vanish. (Namely, that for which  $j=j'$  and  $k=k'$ .) With the normalization of Eq. (7) the value of this non-zero integral is  $m$ , and thus one obtains the result

$$U_{jk}(t) = \frac{1}{m} \iint_{A_s} u(x,y,t) \psi_{jk}(x,y) \mu(x,y) dx dy \quad (9)$$

which permits one to determine the coefficient  $U_{jk}(t)$  which corresponds to the mode shape  $\psi_{jk}(x,y)$  for a given deformation  $u(x,y,t)$ .

## Modal Analysis of System Motions

The deflection of a viscously damped elastic system due to the action of a distributed load  $p(x,y,t)$  is governed by a differential equation of the form

$$\mu \ddot{u} + \gamma \dot{u} + \tilde{K} u = p(x,y,t) \quad (10)$$

where the various terms have the same meaning as in Eq. (1), and  $\gamma$  is an operator or function which involves only the coordinates and which accounts for energy dissipation (viscous damping) in the system.

In order to determine the equations that govern the responses of the modes of the system one may substitute for  $u(x,y,t)$  in Eq. (10) the series given in Eq. (8). If one uses Eq. (4) to replace the terms involving  $\tilde{K}$  by others involving  $\mu$ , one obtains

$$\sum_{j=1}^{\infty} \sum_{k=1}^{\infty} \left( \mu \ddot{U}_{jk} + \gamma \dot{U}_{jk} + \mu \omega_{jk}^2 U_{jk} \right) \psi_{jk}(x,y) = p(x,y,t). \quad (11)$$

Now one may again try to use the strategem that was employed in the derivation of Eq. (9). Namely, one may multiply both sides of Eq. (11) by  $\psi_{j_1 k_1}(x,y)$ , integrate over the entire system, and apply the orthogonality condition. However, in general the damping operator  $\gamma$  and the mass distribution  $\mu$  may not be related in any manner which permits one to apply to the  $\gamma$  terms the orthogonality condition of Eq. (5) which involves  $\mu$ .

Fortunately, the special case where  $\gamma = \beta\mu$ , with  $\beta$  a constant, turns out to be a useful one, and may serve as a good approximation for many kinds of lightly damped systems. In this case the previously discussed strategem does work. Its use in conjunction with the normalization of Eq. (7) yields a result which one may write as

$$\boxed{m \ddot{U}_{jk} + c \dot{U}_{jk} + K_{jk} U_{jk} = F_{jk}(t)} \quad , \quad (12)$$

if one defines



$$c \equiv m \beta \quad K_{jk} \equiv m \omega_{jk}^2 \quad (13)$$

$$F_{jk}(t) \equiv \iint_{A_s} p(x,y,t) \psi_{jk}(x,y) dx dy.$$

For a given elastic system subject to a given loading, the "modal mass"  $m$ , the "modal damping coefficient"  $c$ , and the "modal stiffness"  $K_{jk}$  are constant, whereas the "modal force"  $F_{jk}$  is a function of time only. Equation (12), which governs the modal coefficients  $U_{jk}$ , may be recognized to be the same as the well-known equation that governs the displacement of a linear single-degree-of-freedom system consisting of a mass  $m$ , mounted on parallel arrangement of a spring (with stiffness  $K_{jk}$ ) and dashpot (with viscous damping coefficient  $c$ ), and subject to a force  $F_{jk}(t)$  which varies with time.

It is of interest to note that Eq. (12) applies for each mode (i.e. for all values of  $j$  and  $k$ ), but that the equation for a given  $j$  and  $k$  does not involve terms in other values of  $j$  and  $k$ . There is then no interaction between the modal motions, and the modes are said to be "uncoupled". Such uncoupling does not occur in the most general case, in which orthogonality cannot be applied to the damping terms involving  $\gamma$ , and in which a summation involving all modes would appear in Eq. (12) instead of the single  $c U_{jk}$  term. However, as mentioned previously, one may usually ignore this damping coupling for lightly damped systems.

#### Spatial Averages of Deformations; System Kinetic Energy

If one squares both sides of Eq. (8), multiplies the result by  $\mu$ , and integrates over the entire system, one obtains

$$\begin{aligned} \iint_{A_s} u^2(x,y,t) \mu(x,y) dx dy &= \sum_{j=1}^{\infty} \sum_{k=1}^{\infty} \sum_{j'=1}^{\infty} \sum_{k'=1}^{\infty} \iint_{A_s} U_{jk} U_{j'k'} \psi_{jk} \psi_{j'k'} \mu dx dy \\ &= m \sum_{j=1}^{\infty} \sum_{k=1}^{\infty} U_{jk}^2(t) \end{aligned} \quad (14)$$

where the second form of the right-hand side follows from orthogonality, Eq. (5), and from Eq. (7). The weighted spatial average value of  $u^2$ , with the averaging done over the entire system and with  $\mu$  being used as a weighting function, is defined as

$$\overline{u^2(t)} = \left[ \iint_{A_s} u^2(x,y,t) \mu(x,y) dx dy \right] \cdot \left[ \iint_{A_s} \mu(x,y) dx dy \right]^{-1}. \quad (15)$$

By comparing this expression with Eq. (14) and (7) one finds that

$$\boxed{\overline{u^2(t)} = \sum_{j=1}^{\infty} \sum_{k=1}^{\infty} U_{jk}^2(t)} \quad (16)$$

and thus that the weighted mean square displacement is equal to the sum of the squares of all the modal displacements  $U_{jk}(t)$ . For uniform systems, in which the mass distribution  $\mu$  is constant, the weighted mean-square displacement is equal to the ordinary mean-square displacement, as may easily be verified from Eq. (15); Eq. (16) then applies to the ordinary mean-square displacement.

If one writes an expression like Eq. (14) in terms of the velocity  $v(x,y,t)$  and the corresponding modal velocity coefficients  $V_{jk}$ , then the integral on the left-hand side of that expression may be seen to represent twice the total kinetic energy  $T_T$  of the system. Thus, one may write

$$T_T(t) = \frac{1}{2} \iint_{A_s} v^2(x,y,t) \mu dx dy = \sum_{j=1}^{\infty} \sum_{k=1}^{\infty} T_{jk}(t) \quad (17)$$

where

$$T_{jk}(t) = \frac{m}{2} V_{jk}^2(t) \quad (18)$$



denotes the kinetic energy associated with mode (j,k). Equation (17) shows that the total kinetic energy is equal to the sum of all of the modal kinetic energies, and also to  $\frac{1}{2} m \bar{v}^2$ , where the weighted mean-square velocity  $\bar{v}^2$  is defined analogously to  $\bar{u}^2$  in Eq. (15).

Expressions for the time- and space- averages may be obtained, of course, simply by averaging both sides of Eqs. (16)-(18) with respect to time.

### Summary and Conclusions

The definition of modes of elastic systems and the properties of such modes have been reviewed in rather general terms in the present section.

It has been indicated that any physically reasonable displacement, velocity, etc., distribution can be represented as a superposition of corresponding modal distributions.

Orthogonality of the modes, which applies for systems with elastically constrained boundaries (and thus for all the usually considered ideal cases and for most of those generally of practical interest), has been shown to lead to uncoupled equations of motions for the individual modes, except where interaction occurs due to damping.

The uncoupled modal equations of motion have been demonstrated to be precisely of the same form as those for independent single-degree-of-freedom mass-spring-dashpot systems. Thus, by use of the concept of modes one may replace the complex problem of determining the motions of a continuous distributed elastic system by the much simpler problem of evaluating the motions of single-degree-of-freedom systems which represent the modes of the elastic system. The motions of the total system may then be obtained by superposition of all of the modal motions, spatial averages of motion quantities may easily be determined from the corresponding modal amplitude coefficients, and the total kinetic energy of the system may be found simply by adding all of the modal kinetic energies.

## POWER FLOW BETWEEN TWO MODES

In many practical problems involving the vibrations of complex elastic systems, particularly where the excitations and responses are random, one is more concerned with understanding the interactions of various parts of complex elastic systems and with estimating average vibration levels, than with obtaining more detailed information about the system vibrations. One finds that energy analyses are particularly useful in such cases. These analyses tend to be simple, since they can make use of the easily applied concept of conservation of energy. And, as evident from the previous section, if one knows the kinetic energy of a system one knows its mean-square velocity.

Before concerning oneself with the interaction of two (or more) interconnected elastic systems, where each system contains many modes, one may do well to develop an understanding of the interaction of just two modes. It is clear that in the steady state the vibrational energy of a mode of a two-mode system is affected by the power supplied to the mode from external sources, by the power dissipated by the mode, and also by the power that the mode in question receives from (or supplies to) the other mode. The directly supplied and the dissipated powers can generally be evaluated and related to modal parameters and energies with relative ease. In order to make an energy approach useful, however, one also needs expressions which relate the mode-to-mode power flow to modal energies. It is the purpose of the present section to develop such expressions.

It has been pointed out that a mode of an elastic system behaves precisely like an ideal single-degree-of-freedom system consisting of the modal mass mounted on a spring (having the modal stiffness) and a dashpot (with the modal damping coefficient). Consequently, the energy transfer between two modes is studied here by analyzing the corresponding behavior of two coupled mass-spring-dashpot systems, as shown schematically in Fig. 1. In order to facilitate the analysis, and since linear systems are of primary interest, all of the present discussion is restricted to linear coupling.

The concepts of impedance and admittance are reviewed in the first of the following sections. Thereafter the restrictions which the assumption of conservative coupling places on these quantities are delineated. (Conservative coupling is coupling which neither dissipates nor supplies energy, and thus is of



primary interest where one wishes to study the energy flow between systems.) Next are developed expressions for the modal kinetic energies. Finally, it is demonstrated that the time-average power flow between two modes is proportional to the difference in the time-average modal kinetic energies under some conditions which are often approximated in practice.

### Admittances and Impedances

In order to maintain as much generality as possible, it is useful initially to place no restriction on the coupling element  $L$  of Fig. 1, except for requiring it to be linear. This restriction implies that  $L$  can give rise only to linear terms in the equations of motion, so that response variables (displacements, velocities, etc.) are proportional to excitation variables (forces, rate of change of force, etc.), regardless of any of the amplitudes involved.

The properties of linear systems subject to purely sinusoidal excitations may be studied conveniently by use of phasor notation (Ref. 4). In this notation each sinusoidally varying dynamic variable is represented by a complex quantity or "phasor", and the actual time functions are obtained by taking the real part of the product of the corresponding phasor and  $e^{-i\omega t}$ , where  $\omega$  denotes the radian frequency,  $t$  time, and  $i = \sqrt{-1}$ . Thus, for example, if  $\underline{V}_j$  is the phasor corresponding to the velocity  $V_j(t)$ , then

$$V_j(t) = \text{Re}[\underline{V}_j e^{-i\omega t}] = \text{Re}[\underline{V}_j] \cdot \cos \omega t + \text{Im}[\underline{V}_j] \cdot \sin \omega t \quad (19)$$

In view of the linearity of the two-mass system of Fig. 1 the velocity phasors of the two masses obey the relation

$$\underline{V}_1 = Y_{11} \underline{F}_1 + Y_{12} \underline{F}_2 \quad (20)$$

$$\underline{V}_2 = Y_{21} \underline{F}_1 + Y_{22} \underline{F}_2$$

where  $\underline{F}_1$  and  $\underline{F}_2$  denote phasors corresponding to the forces  $F_1$  and  $F_2$ . The quantities  $Y_{rs}$  are called "admittances"; from Eqs. (20) it is evident that  $Y_{11} = \underline{V}_1/\underline{F}_1$  if  $\underline{F}_2 = 0$ ,  $Y_{12} = \underline{V}_1/\underline{F}_2$  if  $\underline{F}_1 = 0$ ,  $Y_{21} = \underline{V}_2/\underline{F}_1$  if  $\underline{F}_2 = 0$ , etc.

By solving Eqs. (20) for  $\underline{F}_1$  and  $\underline{F}_2$  one finds that one may write

$$\begin{aligned}\underline{F}_1 &= Z_{11} \underline{V}_1 + Z_{12} \underline{V}_2 \\ \underline{F}_2 &= Z_{21} \underline{V}_1 + Z_{22} \underline{V}_2\end{aligned}\tag{21}$$

where

$$\begin{aligned}Z_{11} &= Y_{22}/D, \quad Z_{22} = Y_{11}/D, \quad Z_{12} = -Y_{12}/D, \quad Z_{21} = -Y_{21}/D \\ D &= Y_{11}Y_{22} - Y_{12}Y_{21}.\end{aligned}\tag{22}$$

The  $Z_{rs}$  quantities are called "impedances". From Eqs. (21) one may observe that  $Z_{11} = \underline{F}_1/\underline{V}_1$  if  $\underline{V}_2=0$ ,  $Z_{12} = \underline{F}_1/\underline{V}_2$  if  $\underline{V}_1=0$ , etc. Alternately, by solving Eqs. (21) for  $\underline{V}_1$  and  $\underline{V}_2$  one obtains Eqs. (20) and finds that the admittances may be expressed in terms of the impedances as

$$\begin{aligned}Y_{11} &= Z_{22}/E, \quad Y_{22} = Z_{11}/E, \quad Y_{12} = -Z_{12}/E, \quad Y_{21} = -Z_{21}/E \\ E &= Z_{11}Z_{22} - Z_{12}Z_{21} = 1/D.\end{aligned}\tag{23}$$

The impedances and admittances in general are complex quantities and functions of the frequency  $\omega$ . They depend in general on the uncoupled system parameters, as well as on the coupling element.

### Implications of Conservative Coupling

In order to study the flow of energy between systems it is useful to introduce the restriction of conservative coupling in addition to that of linear coupling; that is, to postulate coupling which neither dissipates nor supplies energy. Only for such coupling is the energy leaving one of two coupled systems equal to the energy reaching the other, and only for such coupling can one speak unambiguously of energy flow from one system to the other. The present section consequently is concerned with the restrictions which are imposed on the various admittances and impedances by the requirement that the coupling be conservative.

Conservation of energy demands that in the steady state the total work done per cycle by the external forces be just equal to the total energy dissipated per cycle. In view of the well-known relations between power and forces and velocities in complex notation, as also summarized in Ref. 5, this statement of energy conservation may be written as

$$\frac{1}{2} \operatorname{Re}[\underline{F}_1 \underline{V}_1^*] + \frac{1}{2} \operatorname{Re}[\underline{F}_2 \underline{V}_2^*] = \frac{1}{2} |\underline{V}_1|^2 \operatorname{Re}[Z_{11}] + \frac{1}{2} |\underline{V}_2|^2 \operatorname{Re}[Z_{22}] , \quad (24)$$

where the asterisk indicates the complex conjugate of the quantity to which it is appended.

Substitution for  $F_1$  and  $F_2$  from Eqs. (21), replacement of  $|\underline{V}_1|^2$  by  $\underline{V}_1 \underline{V}_1^*$ , and elimination of terms common to both sides of the resulting equation, permits one to reduce the foregoing relation to

$$\operatorname{Re}[Z_{12} \underline{V}_2 \underline{V}_1^* + Z_{21} \underline{V}_1 \underline{V}_2^*] = 0 . \quad (25)$$

Since  $\underline{V}_2 \underline{V}_1^* = (\underline{V}_1 \underline{V}_2^*)^*$ , this result implies

$$\begin{aligned} (\operatorname{Re}[Z_{21}] + \operatorname{Re}[Z_{12}]) \operatorname{Re}(\underline{V}_1 \underline{V}_2^*) - \\ (\operatorname{Im}[Z_{21}] - \operatorname{Im}[Z_{12}]) \operatorname{Im}(\underline{V}_1 \underline{V}_2^*) = 0. \end{aligned} \quad (26)$$

But this relation must hold for all  $\underline{V}_1$  and  $\underline{V}_2$ , and thus for all values of  $\operatorname{Re}(\underline{V}_1 \underline{V}_2^*)$  and  $\operatorname{Im}(\underline{V}_1 \underline{V}_2^*)$ . Consequently the coefficients of these two quantities in the foregoing expression must vanish, and

$$\begin{aligned} \operatorname{Re}[Z_{21}] &= - \operatorname{Re}[Z_{12}] \\ \operatorname{Im}[Z_{21}] &= \operatorname{Im}[Z_{12}] \end{aligned} . \quad (27)$$

Conservative coupling thus implies that the "transfer impedances" satisfy

$$Z_{12} = - Z_{21}^* , \quad |Z_{12}| = |Z_{21}| , \quad (28)$$



and in view of Eqs. (22) or (23), that the transfer admittances obey

$$|Y_{12}| = |Y_{21}| \quad . \quad (29)$$

### Time-Average Modal Kinetic Energies

Since it is desired to develop relations which express the power flow between modes in terms of the time-average modal kinetic energies, it is of interest first to study these energies from a rather general viewpoint.

Clearly, the time-average value of the kinetic energy  $T_j$  of a mode  $j$  is given by

$$\langle T_j \rangle = \frac{m_j}{2} \langle V_j^2(t) \rangle \quad (30)$$

where  $m_j$  denotes the modal mass (or the mass of the system of which  $j$  is a mode) and  $V_j$  the modal velocity, and where the brackets  $\langle \dots \rangle$  indicate averaging with respect to time. The time-average value of the square of the modal velocity is also known as the mean-square modal velocity; it is independent of time, in the steady state. The problem of determining the average kinetic energies thus corresponds to that of finding the mean-square velocities, and these may be found from the velocities per se.

If the velocity of mass 1 of Fig. 1 due to a unit impulse force  $\delta(t)$  acting on this mass is given by  $y_{11}(t)$ , then the velocity  $V_1(t)$  due to a general force  $F_1(t)$  acting on mass 1 may be expressed as a convolution (Refs. 1,6):

$$V_1(t) = \int_{-\infty}^{\infty} y_{11}(\alpha) F_1(t-\alpha) d\alpha \quad . \quad (31)$$

Here  $\alpha$  is a dummy time variable, which disappears after the integration is performed and limits are substituted. Similarly, if the velocity of mass 1 due to a unit impulse force acting on mass 2 is given by  $y_{12}(t)$ , then the velocity of mass 1 due to a general force  $F_2(t)$  acting on mass 2 may be written as

$$V_1(t) = \int_{-\infty}^{\infty} y_{12}(\beta) F_2(t-\beta) d\beta \quad , \quad (32)$$

where  $\beta$  is again a dummy variable.

According to the superposition principle, the response produced by a combination of excitations acting on a linear system is equal to the sum of the responses due to the individual excitations. Thus, the velocity  $V_1(t)$  of mass 1 due to the combination of a force  $F_1(t)$  acting on mass 1 and a force  $F_2(t)$  acting on mass 2 may be expressed as

$$V_1(t) = \int_{-\infty}^{\infty} y_{11}(\alpha) F_1(t-\alpha) d\alpha + \int_{-\infty}^{\infty} y_{12}(\beta) F_2(t-\beta) d\beta . \quad (33)$$

The mean-square value of  $V_1(t)$  is obtained by squaring the foregoing equation and averaging over all time.<sup>3</sup> If the two integrals of Eq. (33) are denoted by  $J_{11}=J_{11}(t)$  and  $J_{12}=J_{12}(t)$ , respectively, one may write

$$\langle V_1^2 \rangle = \langle (J_{11}+J_{12})^2 \rangle = \langle J_{11}^2 \rangle + \langle J_{12}^2 \rangle + 2\langle J_{11} J_{12} \rangle . \quad (34)$$

The mixed product term may be expressed as

$$\langle J_{11} J_{12} \rangle = \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} y_{11}(\alpha) y_{12}(\beta) \langle F_1(t-\alpha) F_2(t-\beta) \rangle d\alpha d\beta , \quad (35)$$

where the averaging brackets have been placed only around those quantities which involve the variable  $t$ , with respect to which the averaging is to be performed. If  $F_1$  and  $F_2$  are uncorrelated, that is if

$$\int_{-\infty}^{\infty} F_1(t-\alpha) F_2(t-\beta) dt = 0 \quad (36)$$

for all  $\alpha, \beta$ , then  $\langle F_1(t-\alpha) F_2(t-\beta) \rangle = 0$ , and consequently  $\langle J_{11} J_{12} \rangle = 0$ .

---

<sup>3</sup>In most cases one obtains an adequate approximation by averaging over many periods of the lowest frequency contained in  $V_1(t)$ .



The first term of Eq. (34) may be written analogously to Eq. (35) as

$$\langle J_{11}^2 \rangle = \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} y_{11}(\alpha) y_{11}(\beta) \langle F_1(t-\alpha) F_1(t-\beta) \rangle d\alpha d\beta . \quad (37)$$

Here the term within the averaging brackets may be recognized as the autocorrelation function  $R_1$  of  $F_1(t)$ ; this autocorrelation function obeys the relation (Ref. 6, pp. 20, 32-34)

$$R_1(\alpha-\beta) = \langle F_1(t-\alpha) F_1(t-\beta) \rangle = \langle F_1(t) F_1(t-\alpha+\beta) \rangle =$$

$$\frac{1}{2\pi} \int_{-\infty}^{\infty} S_1(\omega) e^{i\omega(\alpha-\beta)} d\omega \quad (38)$$

where  $S_1$  is defined as the mean square spectral density of the stationary random force  $F_1(t)$  and is related to the autocorrelation function as

$$S_1(\omega) = \int_{-\infty}^{\infty} R_1(\tau) e^{-i\omega\tau} d\tau . \quad (39)$$

The functions  $S_1(\omega)$  and  $R_1(\tau)$  are a Fourier transform pair. So also are the impulse response and frequency response functions  $y_{jk}(t)$  and  $Y_{jk}(\omega)$ , which for all  $j,k$ , obey<sup>4</sup>

---

<sup>4</sup>It should be pointed out that several different valid Fourier transform definitions may be used. These differ only in the coefficients attached to the integrals, but all result in a factor of  $1/2\pi$  for a complete transformation cycle. E.g., some authors use  $1/\sqrt{2\pi}$  in each transformation. Here  $1/2\pi$  has been used where the integrations are performed with respect to  $\omega$  and unity where the integrations are performed with respect to  $\tau$ .

It is also important to note that Eqs. (38)-(40) involve both positive and negative frequencies (and times), whereas only positive frequencies have physical meaning. An experimentally determined spectrum (involving only positive  $\omega$ ) may be converted to a spectrum corresponding to Eqs. (38) and (39) by halving the measured spectrum at each  $\omega$  and plotting the result at both positive and negative numerical values of  $\omega$  (Ref. 12, pp. 11,12).

$$y_{jk}(t) = \frac{1}{2\pi} \int_{-\infty}^{\infty} Y_{jk}(\omega) e^{i\omega t} d\omega, \quad Y_{jk}(\omega) = \int_{-\infty}^{\infty} y_{jk}(\tau) e^{-i\omega\tau} d\tau \quad (40)$$

$$Y_{jk}^*(\omega) = \int_{-\infty}^{\infty} y_{jk}(\tau) e^{i\omega\tau} d\tau.$$

Substitution of Eq. (38) into (37) gives

$$\begin{aligned} \langle J_1^2 \rangle &= \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} y_{11}(\alpha) y_{11}(\beta) S_1(\omega) e^{i\omega(\alpha-\beta)} d\omega d\alpha d\beta \\ &= \int_{-\infty}^{\infty} S_1(\omega) \left[ \int_{-\infty}^{\infty} y_{11}(\alpha) e^{i\omega\alpha} d\alpha \int_{-\infty}^{\infty} y_{11}(\beta) e^{-i\omega\beta} d\beta \right] d\omega \\ &= \int_{-\infty}^{\infty} S_1(\omega) Y_{11}^*(\omega) Y_{11}(\omega) d\omega = \int_{-\infty}^{\infty} S_1(\omega) |Y_{11}(\omega)|^2 d\omega. \end{aligned} \quad (41)$$

Analogously one may write

$$\langle J_{12}^2 \rangle = \int_{-\infty}^{\infty} S_2(\omega) |Y_{12}(\omega)|^2 d\omega \quad (42)$$

where  $S_2(\omega)$  denotes the mean-square spectral density of the force  $F_2(t)$ .

Substitution of the foregoing results into Eq. (34) permits one to write the mean-square velocity  $\langle V_1^2 \rangle$  of mass 1 as

$$\langle V_1^2 \rangle = \int_{-\infty}^{\infty} S_1(\omega) |Y_{11}(\omega)|^2 d\omega + \int_{-\infty}^{\infty} S_2(\omega) |Y_{12}(\omega)|^2 d\omega; \quad (43)$$



and to set down by analogy the mean-square velocity of mass 2 as

$$\langle v_2^2 \rangle = \int_{-\infty}^{\infty} S_1(\omega) |Y_{21}|^2 d\omega + \int_{-\infty}^{\infty} S_2(\omega) |Y_{22}|^2 d\omega \quad (44)$$

#### Calculation of Net Power Flow Between Modes by Superposition of Flows

In order to study the power flow (i.e., the rate of flow of energy) from one mode to another it is convenient to consider one force to act at a time, and then to add the power flows obtained in the two cases to obtain the net power flow. Such addition of power flows is permissible for statistically independent forces  $F_1$  and  $F_2$ , as demonstrated below.

Since the system under consideration is linear, the force  $F_c$  exerted on mass 1 by the coupling element (or by the mass on the coupling element) obeys

$$F_c(t) = \int_{-\infty}^{\infty} h_{11}(\alpha_0) F_1(t-\alpha_0) d\alpha_0 + \int_{-\infty}^{\infty} h_{12}(\beta_0) F_2(t-\beta_0) d\beta_0 \quad (45)$$

in analogy with Eq. (33). Here  $h_{11}$  and  $h_{12}$  represent impulse response functions;  $h_{11}(t)$  denotes the coupling force  $F_c(t)$  resulting from a unit impulse of force acting on mass 1,  $h_{12}(t)$  similarly denotes the  $F_c(t)$  resulting from such an impulse acting on mass 2,  $\alpha_0$  and  $\beta_0$  are dummy time variables.

The time-average power  $P_{12}$  supplied to the coupling element may be obtained by multiplying the instantaneous force as given by Eq. (45) by the instantaneous velocity as given by Eq. (33), and averaging with respect to time. One obtains a result which one may write

$$P_{12} = \langle F_c(t) V_1(t) \rangle = P_{12a} - P_{21b} + \Delta P \quad (46)$$

where

$$\begin{aligned}
P_{12a} &= \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} h_{11}(\alpha_o) y_{11}(\alpha) \langle F_1(t-\alpha_o) F_1(t-\alpha) \rangle d\alpha_o d\alpha \\
P_{21b} &= - \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} h_{12}(\beta_o) y_{12}(\beta) \langle F_2(t-\beta_o) F_2(t-\beta) \rangle d\beta_o d\beta \\
\Delta P &= \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} h_{11}(\alpha_o) y_{12}(\beta) \langle F_1(t-\alpha_o) F_2(t-\beta) \rangle d\alpha_o d\beta \\
&\quad + \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} h_{12}(\beta_o) y_{11}(\alpha) \langle F_2(t-\beta_o) F_1(t-\alpha) \rangle d\beta_o d\alpha .
\end{aligned} \tag{47}$$

For conservative coupling, the time-average power flowing into the coupling element from mode 1 at steady state is exactly equal to the power flowing into mode 2, so that  $P_{12}$  of Eq. (46) represents also the power flow from mode 1 to mode 2. If  $F_1(t)$  and  $F_2(t)$  are uncorrelated, so as to satisfy Eq. (36), then the time averages appearing in the integrands of the foregoing expression for  $\Delta P$  vanish and  $\Delta P=0$ . Since  $P_{12a}$  depends only on the excitation  $F_1(t)$ , it may be interpreted as the power flow from mode 1 to mode 2 due to the action of  $F_1(t)$  by itself (i.e., with  $F_2=0$ ).  $P_{21b}$  may be similarly interpreted, except that the minus sign included in its definition changes the direction considered that of positive power flow;  $P_{21b}$  thus represents the power flow from mode 2 to mode 1 due to the action of  $F_2(t)$  by itself.

In view of the foregoing discussion one may thus write the net time-average (steady-state) power flow from mode 1 to mode 2 as

$$P_{12} = P_{12a} - P_{21b} \tag{48}$$

where now  $P_{12a}$  denotes the flow from mode 1 to mode 2 if only mode 1 is excited,  $P_{21b}$  denotes the flow from mode 2 to mode 1 if only mode 2 is excited, and where the two excitations are taken to be uncorrelated.



One may calculate the power flow between modes directly by carrying out the integrations indicated in Eqs. (47), as Scharton has in essence done for a particular kind of coupling (Ref. 7). However, one may retain a greater degree of generality and present a discussion which is relatively easily understood by taking an approach based on Eq. (48) and on energy balance arguments, as in the following paragraphs.

#### General Expression for Power Flow

If only a force  $F_1$  is present (that is, if  $F_2=0$ ), then mass 2 is excited only through the coupling element, and in the steady state all of the power dissipated by the damping of system 2 must be supplied to this system from system 1 via the coupling element. An analogous statement applies if  $F_1=0$  and  $F_2 \neq 0$ . Let the time-average power flow from system  $j$  to system  $k$  be denoted by  $P_{jk}$ , and let the added subscript "a" designate quantities corresponding to the  $F_2=0$  case, and subscript "b" to the  $F_1=0$  case. In view of the previous energy balance arguments and of Eqs. (43) and (44) one may write<sup>5</sup>

$$\begin{aligned} P_{12a} &= c_2 \langle v_{2a}^2 \rangle = c_2 \int_{-\infty}^{\infty} S_1(\omega) |Y_{21}|^2 d\omega \\ P_{21b} &= c_1 \langle v_{1b}^2 \rangle = c_1 \int_{-\infty}^{\infty} S_2(\omega) |Y_{12}|^2 d\omega \end{aligned} \quad (49)$$

where  $S_1$  and  $S_2$  denote the mean square spectral densities of the forces  $F_1$  and  $F_2$ . In view of Eqs. (48) and (49), the net power flow from system 1 to 2, when both external forces are acting, obeys the equation

$$P_{12} = \int_{-\infty}^{\infty} [c_2 S_1(\omega) - c_1 S_2(\omega)] |Y_{12}|^2 d\omega. \quad (50)$$

<sup>5</sup>Note that  $c_1 = \text{Re}[Z_{11}]$  for the system of Fig. 1, if the coupling is conservative. In view of the definition of  $Z_{11}$ , as indicated following Eq. (22),  $Z_{11}$  is measured with  $V_2=0$ ; hence no (dissipative) elements in system 2 can contribute terms to  $Z_{11}$ . Since the coupling is conservative it cannot contribute real terms to  $Z_{11}$ . Analogously,  $c_2 = \text{Re}[Z_{22}]$ .

### Relation Between Power Flow and Kinetic Energies

If  $F_1$  and  $F_2$  are "white noise" forces, that is if  $S_1(\omega) = S_1$  and  $S_2(\omega) = S_2$  are constant (independent of frequency), then one may introduce the notation

$$I_{jk} \equiv \int_{-\infty}^{\infty} |Y_{jk}|^2 d\omega \quad (51)$$

and rewrite Eqs. (33) and (34) as

$$\begin{aligned} \langle v_1^2 \rangle &= S_1 I_{11} + S_2 I_{12} \\ \langle v_2^2 \rangle &= S_1 I_{21} + S_2 I_{22} \end{aligned} \quad (52)$$

and Eq. (50) as

$$P_{12} = I_{12}(c_2 S_1 - c_1 S_2) \quad (53)$$

By solving Eqs. (52) for  $S_1$  and  $S_2$  in terms of  $\langle v_1^2 \rangle$  and  $\langle v_2^2 \rangle$  and introducing the result into Eq. (53), one obtains an expression for the power flow  $P_{12}$  in terms of the mean square velocities. By expressing the mean-square velocities  $\langle v_j^2 \rangle$  in terms of the time-average kinetic energies  $\langle T_j \rangle$  as indicated in Eq. (30), one finds that one may then write

$$P_{12} = \phi_{12} \langle T_1 \rangle - \phi_{21} \langle T_2 \rangle \quad (54)$$

where

$$\begin{aligned} \phi_{12} &\equiv \frac{2 I_{12} \left( \frac{c_2}{m_1} I_{22} + \frac{c_1}{m_1} I_{12} \right)}{I_{11} I_{22} - I_{12}^2} \\ \phi_{21} &\equiv \frac{2 I_{12} \left( \frac{c_1}{m_2} I_{11} + \frac{c_2}{m_2} I_{12} \right)}{I_{11} I_{22} - I_{12}^2} \end{aligned} \quad (55)$$



and where one has made use of the fact that  $I_{12}=I_{21}$ , as implied by Eq. (29).

#### Power Flow for Weak Coupling

The two systems of Fig. 1 may be said to be weakly coupled if the transfer admittance integrals  $I_{12}=I_{21}$  have values which are much smaller than those of the "self-admittance" integrals  $I_{11}$  and  $I_{22}$ , and if the transfer impedances and admittances  $Z_{12}$ ,  $Z_{21}$ ,  $Y_{12}$ ,  $Y_{21}$ , do not affect the values of  $I_{11}$  and  $I_{22}$  significantly.

Then, substitution of the first of Eqs. (23) into Eq. (51) yields

$$I_{11} = \int_{-\infty}^{\infty} |Y_{11}|^2 d\omega = \int_{-\infty}^{\infty} \left| \frac{Z_{22}}{Z_{11}Z_{22} - Z_{12}Z_{21}} \right|^2 d\omega \approx \int_{-\infty}^{\infty} |Z_{11}|^{-2} d\omega \quad (56)$$

$$\approx \int_{-\infty}^{\infty} \left| m_1(i\omega) + c_1 + k_1(i\omega)^{-1} \right|^{-2} d\omega = \frac{\pi}{c_1 m_1} .$$

Note that the approximations indicated in Eq. (56) apply even if the transfer impedances and admittances are not strictly negligible (as compared to the self terms) over the entire frequency range, provided that the value of the integral is not significantly affected by contributions corresponding to transfer term values which are not negligible. For example, if  $Z_{11}$  and/or  $Z_{22}$  are small compared to  $Z_{12}$  and/or  $Z_{21}$  in certain frequency ranges (due to resonances of masses 1 and/or 2), then the approximations of Eq. (56) still hold, provided that these frequency ranges are narrow enough.

By use of the previously stated requirement  $I_{12} \ll I_{11}$ ,  $I_{22}$  for loose coupling and of Eq. (56) one finds that one may restate Eqs. (54) and (55) as

$$\phi_{12} = \phi_{21} \approx \frac{2}{\pi} c_1 c_2 I_{12} \equiv \phi_c , \quad (57)$$

$$P_{12} = \phi_c (\langle T_1 \rangle - \langle T_2 \rangle) .$$

## Power Flow for Stiffness and Gyroscopic Coupling

If the coupling element consists of an ideal stiffness (represented by a spring constant  $k_c$ ) and an ideal gyroscopic element<sup>6</sup> (represented by a force/velocity coefficient  $c_g$ ), one finds that

$$\begin{aligned} Z_{11} &= m_1(i\omega) + c_1 + (k_1 + k_c)(i\omega)^{-1} \\ Z_{22} &= m_2(i\omega) + c_2 + (k_2 + k_c)(i\omega)^{-1} \\ Z_{12} &= c_g - k_g(i\omega)^{-1} \\ Z_{21} &= -c_g - k_g(i\omega)^{-1} \end{aligned} \quad (58)$$

One may substitute these impedances into Eqs. (23) to find the various corresponding admittances, and one may then evaluate the integrals  $I_{11}$ ,  $I_{22}$ ,  $I_{12}$  exactly (by use of the theorem of residues, as summarized in Ref. 9, for example, or from the tabulated expression given on p. 72 of Ref. 6). After substitution of the results into Eqs. (55) and a considerable amount of algebraic manipulation one finds that the two coefficients  $\phi_{12}$  and  $\phi_{21}$  are here exactly equal, regardless of the magnitude of the coupling, and are given by

$$\phi_{12} = \phi_{21} = \phi_c = \frac{2}{m_1 m_2} \frac{k_c^2 \alpha_k + c_g^2 \alpha_c}{\alpha_k \alpha_c + (\omega_2^2 - \omega_1^2)^2} \quad , \quad (59)$$

---

<sup>6</sup>A gyroscopic element is defined as one which produces a negative force on mass 2 due to a positive velocity of mass 1, if it results in a positive force on mass 1 due to a positive velocity of mass 2. (See Ref. 8.) Note that a viscous damping element results in forces of the same sign on both masses due to positive velocities, and that gyroscopic coupling is conservative since the  $Z_{12}$  and  $Z_{21}$  of Eqs. (58) satisfy the requirement of Eq. (28).



where

$$\begin{aligned}\alpha_k &\equiv \frac{c_1}{m_1} + \frac{c_2}{m_2} = 2(\xi_1\omega_1 + \xi_2\omega_2) \\ \alpha_c &\equiv \frac{c_1k_2 + c_2k_1}{m_1m_2} = 2\omega_1\omega_2(\xi_1\omega_2 + \xi_2\omega_1) \\ \omega_1^2 &\equiv (k_1+k_c)/m_1, \quad \omega_2^2 \equiv (k_2+k_c)/m_2 \\ \xi_1 &\equiv 2m_1\omega_1, \quad \xi_2 \equiv 2m_2\omega_2.\end{aligned}\tag{60}$$

The definitions of Eqs. (60) were chosen so that  $\omega_j$  represents the natural frequency of the  $j^{\text{th}}$  mass (or mode) if the other mass is kept from moving, and so that  $\xi_j$  denotes the ratio of the damping coefficient  $c_j$  to the critical damping coefficient  $2m_j\omega_j$  of the  $j^{\text{th}}$  mass.

The foregoing result is found to agree with that which Scharton (Ref. 7) derived for pure spring coupling by calculating explicitly the power flow through the coupling element. For small coupling the foregoing result also reduces to that which was obtained by Lyon and Maidanik (Ref. 10) by means of an explicit power flow calculation based on small coupling approximations.

#### Finite Frequency Band Excitation

The previous discussion was based on force spectra  $S_1$  and  $S_2$  that are assumed constant for all frequencies. The results obtained there similarly apply to power flow and system energies involving the entire infinite frequency band. However, a little reflection permits one to interpret these results also for cases where only a finite frequency band is of interest.

One may obtain expressions pertaining to only a finite frequency band by replacing the infinite limits appearing in the integrals of Eqs. (49)-(51) by the frequencies limiting the band. Equations (52)-(55) then still apply, although the various  $I_{jk}$  integrals that enter them now are different.

The magnitude of the admittance  $Y(\omega)$  of a not too highly damped single-degree-of-freedom system is much greater near the system natural frequency (resonance) than at other frequencies, so that the greatest contribution to the value of the integral  $I = \int_{-\infty}^{\infty} |Y(\omega)|^2 d\omega$  is due to the frequency region near the system resonance. Consequently, the value of the integral will be changed only slightly if the infinite limits are replaced by finite ones, provided that the interval of integration (i.e., the frequency band under consideration) still encompasses the system resonance. (See, for example, Ref. 11, or pp. 21ff, 29ff of Ref. 5.)

Similar reasoning applied to a two-degree-of-freedom system (made up of two coupled single-degree-of-freedom systems), which has two resonances, leads one to conclude that the results of Eqs. (56), (57), (59) and (60) also are good approximations if the excitation frequency bands are finite, but encompass both system resonances.

In view of the previously discussed predominance of the responses near the resonances, the aforementioned results may also be taken to be good approximations for cases where the excitation spectra are not flat. All that is required for these approximations to hold is that the excitation spectra exhibit no violent peaks; i.e., no peaks that are of the same order as the peaks in the  $|Y|^2$  curves.

### Summary and Conclusions

It has been shown in this section that the steady-state time-average power flow from one mode to another is proportional to the difference between the time-average kinetic energies of the two modes, provided that both

- 1) the coupling between the two modes is
  - linear (giving rise to a linear differential equation),
  - conservative (neither supplying nor extracting mechanical energy), and
  - light and/or purely spring-like and/or gyroscopic;

and

- 2) the forces acting on the two modes are uncorrelated and have spectra that are relatively flat (as compared to the system admittance spectra) within the frequency band encompassed by the resonances of the coupled system.



Expressions which give the constant of proportionality  $\phi_c$  between power flow and kinetic energy difference in terms of the characteristics of the modes and of the coupling elements have been derived. These expressions are given in Eqs. (57) and (59).

Since the square of the difference between the squares of the modal resonance frequencies appears in the denominator of Eq. (59), one may note that the proportionality constant  $\phi_c$  decreases rapidly as the frequency difference increases. Thus, the power flow between two modes whose resonance frequencies are nearly the same will be much greater than the flow between similar modes whose resonances are widely separated.

The relation between power flow and kinetic energies, which has been established in this section, permits one to analyze the average dynamic behavior of coupled modes rather simply on the basis of energy balance considerations. For such energy analyses it is convenient to visualize coupled modes as shown in Fig. 2. There the modes are represented as storage elements (containing kinetic energies  $\langle T_1 \rangle$  and  $\langle T_2 \rangle$ ), connected by power flow  $P_{12}$ , supplied with power inputs  $A_1$  and  $A_2$  from sources external to the system, and experiencing power losses (dissipations)  $D_1$  and  $D_2$ , respectively.

## POWER FLOW BETWEEN TWO SETS OF MODES

As has been discussed earlier, one may analyze the vibrations of an elastic system in terms of the vibrations of the set of the modes of the system. The interaction between two elastic systems may correspondingly be studied by investigating the interaction between two sets of modes which represent the two systems.

The modes of an ideal system vibrate independently of each other, as has been pointed out previously, and the modes of lightly damped realistic systems tend to behave similarly. Hence, the present section deals with the power flow between two sets of modes, where the members of each set are not coupled to each other, but are coupled to all members of the other set.

The problem discussed in the present section may be visualized with the aid of Fig. 3, which schematically shows two sets of modes, one designated by " $\alpha$ ", the other by " $\beta$ ". It is assumed that no power flow occurs between modes which are members of the same set<sup>7</sup>, but that each mode of the  $\alpha$  set is coupled to each mode of the  $\beta$  set. It is desired to determine the flow of energy from the  $\alpha$  set to the  $\beta$  set in terms of the previously studied mode-to-mode flows, under conditions where each of the modes of each set may dissipate energy and may be supplied with energy from outside the system (as indicated in Fig. 3).

In particular, it is desired to determine under what conditions one may obtain relations for the total net power flow between two sets of modes in the form

$$P_{\alpha\beta} = \phi_{\alpha\beta} T_{\alpha} - \phi_{\beta\alpha} T_{\beta} \quad , \quad (61)$$

in analogy to Eq. (54), which pertains to power flow between two single modes. Here  $T_{\alpha}$  and  $T_{\beta}$  denote the set-averages of the time-average modal kinetic energies, which are discussed in more detail in the subsequent paragraphs. Moreover, one desires to establish for coupled sets a power flow diagram like Fig. 4, in analogy to the power flow diagram of Fig. 2, which pertains to two coupled modes.

Throughout the following discussion it is assumed that the power flow between each pair of coupled modes is proportional to the difference between the time-average modal kinetic energies, as indicated by the second of Eqs. (57). It is convenient for the present purposes to rewrite the aforementioned expression as

---

<sup>7</sup>No power flow between these modes occurs either if these modes are not coupled to each other, or if all of these modes contain equal kinetic energies. See Eq. (57).



$$P_{\alpha j, \beta k} = \phi_{\alpha j, \beta k} (T_{\alpha j} - T_{\beta k}) \quad (62)$$

Here  $P_{\alpha j, \beta k}$  denotes the power flow from the  $j^{\text{th}}$  mode of the  $\alpha$  set to the  $k^{\text{th}}$  mode of the  $\beta$  set, and  $T_{\alpha j}$  and  $T_{\beta k}$  denote the time-average kinetic energies of these two modes (due to effectively white-noise excitation), in agreement with Eq. (57). In Eq. (62) and subsequently the brackets previously used to designate time-averages are omitted for the sake of economy of notation; however, all kinetic energy and power flow quantities appearing here and henceforth are intended to represent time-average values.

#### Power Flow Between Mode Sets in Terms of Set-Average Energies

The total power flow  $P_{\alpha\beta}$  from mode set  $\alpha$  to mode set  $\beta$  is the sum of all the individual mode-to-mode power flows. If set  $\alpha$  contains  $N_\alpha$  modes and set  $\beta$  contains  $N_\beta$  modes, then one may express the set-to-set power flow as

$$P_{\alpha\beta} = \sum_{j=1}^{N_\alpha} \sum_{k=1}^{N_\beta} P_{\alpha j, \beta k} = \sum_{j=1}^{N_\alpha} \sum_{k=1}^{N_\beta} \phi_{\alpha j, \beta k} (T_{\alpha j} - T_{\beta k}) \quad (63)$$

in view of Eq. (62).

The set-average modal kinetic energies  $T_\alpha$  and  $T_\beta$  of the two sets are defined as

$$T_\alpha \equiv \frac{1}{N_\alpha} \sum_{j=1}^{N_\alpha} T_{\alpha j} \quad , \quad T_\beta \equiv \frac{1}{N_\beta} \sum_{k=1}^{N_\beta} T_{\beta k} \quad (64)$$

If Eqs. (61) and (63) are to agree, then the relation

$$\sum_{j=1}^{N_\alpha} \sum_{k=1}^{N_\beta} \phi_{\alpha j, \beta k} (T_{\alpha j} - T_{\beta k}) = \frac{\phi_{\alpha\beta}}{N_\alpha} \sum_{j=1}^{N_\alpha} T_{\alpha j} - \frac{\phi_{\beta\alpha}}{N_\beta} \sum_{k=1}^{N_\beta} T_{\beta k} \quad (65)$$

must hold for all values of  $T_{\alpha j}$  and  $T_{\beta k}$ . Consequently the set-to-set coupling factors must obey

$$\phi_{\alpha\beta} = \left[ N_{\alpha} \sum_{j=1}^{N_{\alpha}} \sum_{k=1}^{N_{\beta}} \phi_{\alpha j, \beta k} T_{\alpha j} \right] \div \left[ \sum_{j=1}^{N_{\alpha}} T_{\alpha j} \right] \quad (66)$$

$$\phi_{\beta\alpha} = \left[ N_{\beta} \sum_{j=1}^{N_{\alpha}} \sum_{k=1}^{N_{\beta}} \phi_{\alpha j, \beta k} T_{\beta k} \right] \div \left[ \sum_{k=1}^{N_{\beta}} T_{\beta k} \right]$$

and the equivalent relations

$$\phi_{\alpha\beta} = N_{\alpha} N_{\beta} \left\{ \frac{1}{N_{\beta}} \sum_{k=1}^{N_{\beta}} \left[ \left( \sum_{j=1}^{N_{\alpha}} \phi_{\alpha j, \beta k} T_{\alpha j} \right) \div \left( \sum_{j=1}^{N_{\alpha}} T_{\alpha j} \right) \right] \right\} \quad (67)$$

$$\phi_{\beta\alpha} = N_{\alpha} N_{\beta} \left\{ \frac{1}{N_{\alpha}} \sum_{j=1}^{N_{\alpha}} \left[ \left( \sum_{k=1}^{N_{\beta}} \phi_{\alpha j, \beta k} T_{\beta k} \right) \div \left( \sum_{k=1}^{N_{\beta}} T_{\beta k} \right) \right] \right\}.$$

The latter expressions have an interesting interpretation. The term within the large square brackets of the first of the foregoing equations is the weighted average of the  $\phi_{\alpha j, \beta k}$  coefficients (for a given value of  $k$ ), with the modal energies  $T_{\alpha j}$  acting as weighting factors. The term within the curved brackets may be seen to be the average (over all values of  $k$ ) of the aforementioned weighted averages. Thus,  $\phi_{\alpha\beta}$  is  $N_{\alpha} N_{\beta}$  times the average value of  $\phi_{\alpha j, \beta k}$  (in the previously indicated sense); this implies that the power flow between mode sets is equal to the average mode-to-mode flow times the number of mode pairs between which power flow occurs.



### Modal Energy-Independent Coupling Factors

If the set-to-set coupling factors  $\phi_{\alpha\beta}$  and  $\phi_{\beta\alpha}$  depend on the modal energies  $T_{\alpha j}$  and  $T_{\beta k}$  as in Eqs. (66) and (67), then use of these factors in Eq. (61) does not simplify the calculation of power flow between mode sets. However, in some cases the coupling factors do turn out to be independent of the energies  $T_{\alpha j}$  and  $T_{\beta k}$ 's, and then Eq. (61) does present a convenient means for calculating the set-to-set power flows. Conditions under which the coupling factors tend to be independent of the energies are explored in the present section.

#### Uniform Coupling

For the case where all of the mode-to-mode coupling factors are equal, that is, where  $\phi_{\alpha j, \beta k} = \phi_0$  for all  $j$  and  $k$ , one finds that the set-to-set coupling coefficients of Eqs. (66) or (67) are independent of the modal kinetic energies and are given by

$$\phi_{\alpha\beta} = \phi_{\beta\alpha} = N_{\alpha}N_{\beta} \phi_0 \quad . \quad (68)$$

One may observe, as also previously mentioned, that here the set-to-set coupling coefficient is  $N_{\alpha}N_{\beta}$  times the mode-to-mode coefficient, and that  $N_{\alpha}N_{\beta}$  is the number of mode-to-mode paths along which power can flow from members of one set to members of the other.

Equation (57) indicates that the assumption of uniform coupling may be useful under some often-encountered practical circumstances. If all modes of set "α" are similar to each other (i.e., have nearly the same resonance frequency and the same amount of damping), if all modes of set "β" are similar to each other, and if all α modes are similarly coupled to all β modes (i.e., if all mode-to-mode coupling is characterized by the same kinds and magnitudes of coupling elements), then all the mode-to-mode coupling factors will be the same. Thus, Eq. (68) applies to mode sets that satisfy the foregoing similarity criteria. Indeed, an essential step in the practical application of the statistical energy approach consists of organizing the set of all modes of a system into subsets within which all modes have similar characteristics, so that relations like Eq. (68) can be applied to these subsets.

### Uniform Modal Energies

If all of the modal kinetic energies in set "α" are equal, and if the same is true of set "β", so that one may write  $T_{\alpha j} = T_{\alpha 0}$  for all j and  $T_{\beta k} = T_{\beta 0}$  for all k, then Eqs. (66) or (67) are found to reduce to

$$\phi_{\alpha\beta} = \phi_{\beta\alpha} = \sum_{j=1}^{N_{\alpha}} \sum_{k=1}^{N_{\beta}} \phi_{\alpha j, \beta k} \quad , \quad (69)$$

This expression indicates that here the set-to-set coupling factor is equal to the sum of the mode-to-mode factors for all coupled mode pairs. (This sum clearly is equal to  $N_{\alpha}N_{\beta}$  times the average value of the mode-to-mode coupling factor between all "α" and "β" modes.)

In order to investigate the circumstances under which the assumption of uniform modal energies is a realistic one, one needs to study the general modal kinetic energy  $T_{\beta k}$ . An expression for this energy may be obtained from a steady-state time-average energy balance relation for the  $k^{\text{th}}$  mode of set "β", which relation may be written as

$$A_{\beta k} + \sum_{j=1}^{N_{\alpha}} P_{\alpha j, \beta k} = D_{\beta k} \quad , \quad (70)$$

or

$$A_{\beta k} + \sum_{j=1}^{N_{\alpha}} \phi_{\alpha j, \beta k} (T_{\alpha j} - T_{\beta k}) = c_{\beta k} \frac{2}{m_{\beta k}} T_{\beta k} \quad . \quad (71)$$

As indicated in Fig. 3,  $A_{\beta k}$  represents the power supplied to the mode directly from external sources,  $D_{\beta k}$  denotes the power dissipated by the mode (which depends on the modal viscous damping coefficient  $c_{\beta k}$  and on the mean square velocity, and thus on the modal kinetic energy  $T_{\beta k}$ ), and  $P_{\alpha j, \beta k}$  denotes the power supplied from the  $j^{\text{th}}$  mode of the "α" set to the mode under consideration.



Solution of Eq. (71) for  $T_{\beta k}$  yields

$$T_{\beta k} = \left( A_{\beta k} + \sum_{j=1}^{N_{\alpha}} \phi_{\alpha j, \beta k} T_{\alpha j} \right) \div \left( \frac{2c_{\beta k}}{m_{\beta k}} + \sum_{j=1}^{N_{\alpha}} \phi_{\alpha j, \beta k} \right), \quad (72)$$

which expresses the kinetic energy  $T_{\beta k}$  of the single mode  $\beta k$  in terms of the modal and coupling parameters and of the energies  $T_{\alpha j}$  of all of the modes of the " $\alpha$ " set. One may use Eq. (72) to investigate under what conditions all of the " $\beta$ " modes possess approximately equal time-average kinetic energies.

If those terms of Eq. (72) which contain the coupling factors are negligible compared to the others, then this equation reduces to

$$T_{\beta k} \approx A_{\beta k} m_{\beta k}^{1/2} c_{\beta k} \quad . \quad (73)$$

For negligible coupling the power input may be shown to be given by<sup>8</sup>

---

<sup>8</sup>If coupling effects are negligible, then the power input is equal to the power dissipation, or (omitting all subscripts since only one system is being considered here)

$$A = D = c \langle V^2 \rangle = cSI = \pi S/m \quad ,$$

in view of Eqs. (52) and (56).

One may verify that one may obtain the same result by direct calculation of the power input:

$$A = \langle F(t) V(t) \rangle = \int_{-\infty}^{\infty} y(\alpha) \langle F(t-\alpha) F(t) \rangle d\alpha \quad .$$

Since the force autocorrelation function  $R_f(\alpha)$  is related to the spectrum  $S$  of a white noise force (Ref. 12) as

$$\langle F(t-\alpha) F(t) \rangle = R_f(\alpha) = 2\pi S_a \delta(\alpha) \quad ,$$

where  $\delta(\alpha)$  is the delta or unit impulse function, the above reduces to

$$A = 2\pi S \int_{-\infty}^{\infty} y(\alpha) \delta(\alpha) d\alpha = \pi S [y(0+) + y(0-)] = \pi S/m \quad ,$$

since  $y(0+) = 1/m$ ,  $y(0-) = 0$ .

$$A_{\beta k} = \pi S_{\beta k} / m_{\beta k} \quad , \quad (74)$$

where  $S_{\beta k}$  denotes the spectral density of the force which acts directly on the  $\beta k$  mode. One may then rewrite Eq. (74) as

$$T_{\beta k} \approx (\pi/2)(S_{\beta k}/c_{\beta k}) \quad . \quad (75)$$

This result indicates that in the presence of small coupling the kinetic energies of all " $\beta$ " modes will be equal if the excitations are adjusted so that the ratio of modal force spectral density to modal damping is the same for all modes. As a special case, if all modes of the  $\beta$  set are equally excited and equally damped, then they will have equal kinetic energies.

Some other observations relevant to uniform distribution of energies may be made from an analysis of indirectly excited modes. If the modes of the " $\beta$ " set are lightly damped and not directly excited ( $A_{\beta k}=0$ ) then the net power flow to the  $j^{\text{th}}$  mode from all the modes of the " $\alpha$ " set must be very small, and one may determine from Eq. (71) or Eq. (72) that

$$T_{\beta k} \approx \left( \sum_{j=1}^{N_{\alpha}} \phi_{\alpha j, \beta k} T_{\alpha j} \right) \div \left( \sum_{j=1}^{N_{\alpha}} \phi_{\alpha j, \beta k} \right) \quad (76)$$

Equation (76) shows  $T_{\beta k}$  here to be a weighted average of all of the  $T_{\alpha j}$ 's (with the coupling coefficients  $\phi_{\alpha j, \beta k}$  acting as weighting factors). One notes that the kinetic energies  $T_{\beta k}$  of all of the " $\beta$ " modes will be equal if the sums appearing in Eq. (76) are independent of the index  $k$ , or if all of the " $\alpha$ " modes have the same kinetic energy  $T_{\alpha j}$ . Thus, if the total coupling to all " $\beta$ " modes is uniform, and/or if the energy distribution in the " $\alpha$ " set is uniform, then the energy distribution in the indirectly excited " $\beta$ " set will also be uniform.

Equation (76) also implies that the distribution of energies  $T_{\beta k}$  in the indirectly excited " $\beta$ " set will generally be smoother than that of the energies  $T_{\alpha j}$  in the " $\alpha$ " set, if the summations do not depend very strongly on the index  $k$ . In other words, the  $T_{\beta k}$ 's will approach equality to each other more closely than do the  $T_{\alpha j}$ 's, provided that the coupling of the " $\beta$ " modes to all of the " $\alpha$ " modes is about the same for all of the " $\beta$ " modes.



One may thus visualize the behavior of a cascade of lightly damped mode sets, where only the first set is directly excited, where each set is relatively loosely coupled to its neighbors, and where the mode-to-mode coupling for each pair of adjacent sets is more or less uniform. (This does not imply that the mode-to-mode coupling need have the same value for all set pairs.) If the first set of such a cascade is excited so that an arbitrary distribution of kinetic energies is produced in it, then the second set will attain an energy distribution which is more uniform than that in the first, in accordance with the previous paragraph. Similarly, the energy distribution in the third set will then be more uniform than that in the second, and so on.

#### Power Flow in the Ensemble Average

In practice one rarely deals with structures whose every detail is precisely known. For example, manufacturing tolerances produce geometric differences, and variations in material properties result in deviations from ideal homogeneity and isotropy. Hence, one may expect to encounter some differences in the properties and distributions of modes in nominally identical structures, and thus in the responses of these structures to the same excitation. In order to study the behavior of "average" structures, one is led to consider the power flow that may be expected to occur "on the average" between two elastic systems, where each of the two is a member of an ensemble, and where the members of each ensemble can differ from each other to some degree.

Instead of dealing with only a single " $\alpha$ " set and a single " $\beta$ " set of modes (corresponding to a single  $\alpha$  elastic system and a single  $\beta$  elastic system), it is necessary here to deal with collections or "ensembles" of such sets (corresponding to ensembles of elastic systems). The problem to be considered here consists of determining the ensemble average of the time-average power flow between  $\alpha$  and  $\beta$  mode sets, in terms of the ensemble average of the time-average kinetic energies.

Consider an "experiment", where one selects at random from the ensembles one set of the type  $\alpha$  and one of the type  $\beta$ , subjects each of the selected pair of sets to a specified excitation, and measures the time-average power flow and kinetic energies. It is desired to express the average power flow measured in a number of such experiments in a form analogous to Eq. (61), as

$$\bar{P}_{\alpha\beta} = \tilde{\varphi}_{\alpha\beta} \bar{T}_{\alpha} - \tilde{\phi}_{\beta\alpha} \bar{T}_{\beta} \quad , \quad (77)$$

where the bar denotes the average of the values obtained in these experiments (i.e., the ensemble average) and the tilde ( $\sim$ ) is intended to differentiate the present coupling coefficients from those of Eq. (61), which pertain to two fully defined mode sets; i.e. to a single "experiment".

One may similarly modify Eqs. (62)-(66) to apply to ensemble averages and to the coupling coefficients as defined by Eq. (77) and arrive at expressions analogous to Eqs. (67). Noting that one may interchange the order of summation and averaging (which essentially is another summation), one obtains

$$\begin{aligned} \tilde{\varphi}_{\alpha\beta} &= N_{\alpha}N_{\beta} \left\{ \frac{1}{N_{\beta}} \sum_{k=1}^{N_{\beta}} \left[ \left( \sum_{j=1}^{N_{\alpha}} \overline{\phi_{\alpha j, \beta k} T_{\alpha j}} \right) \div \left( \sum_{j=1}^{N_{\alpha}} \overline{T_{\alpha j}} \right) \right] \right\} \\ \tilde{\phi}_{\beta\alpha} &= N_{\alpha}N_{\beta} \left\{ \frac{1}{N_{\alpha}} \sum_{j=1}^{N_{\alpha}} \left[ \left( \sum_{k=1}^{N_{\beta}} \overline{\phi_{\alpha j, \beta k} T_{\beta k}} \right) \div \left( \sum_{k=1}^{N_{\beta}} \overline{T_{\beta k}} \right) \right] \right\}. \end{aligned} \quad (78)$$

The average of the product of two quantities is not in general equal to the product of the averages; one cannot in general replace  $\overline{\phi_{\alpha j, \beta k} T_{\alpha j}}$  by  $(\phi_{\alpha j, \beta k})(\overline{T_{\alpha j}})$ , as one would like to do in order to simplify Eqs. (78). However, the aforementioned interchange of the averaging and multiplication processes is valid for random variables that are statistically independent. This means that  $\overline{\phi_{\alpha j, \beta k} T_{\alpha j}} = (\phi_{\alpha j, \beta k})(\overline{T_{\alpha j}})$ , provided that the probability of obtaining any value of  $T_{\alpha j}$  in a given "experiment" is unaffected by the value of  $\phi_{\alpha j, \beta k}$  in that experiment, and vice versa.

The various modal kinetic energies which are produced by a given excitation distribution do depend on the various mode-to-mode coupling factors (as one can readily determine from any energy balance analysis), and thus cannot be entirely statistically independent of the coupling. However, it is likely that a given  $T_{\alpha j}$  will depend only very weakly on a single  $\phi_{\alpha j, \beta k}$ , particularly if many modes are present in the two sets; then these two quantities may be very nearly independent. If the excitation distribution may also vary from experiment to experiment (e.g., in the relatively realistic case where the modal excitations depend on the sample structure selected for an "experiment") then statistical independence is even more likely.



For statistically independent mode-to-mode coupling coefficients and modal kinetic energies one may replace  $\phi_{\alpha j, \beta k} \bar{T}_{\alpha j}$  by  $(\phi_{\alpha j, \beta k}) \cdot \bar{T}_{\alpha j}$  in the first of Eqs. (78) and perform a similar replacement in the second of these equations. If  $\bar{T}_{\alpha j}$  is independent of the mode number  $j$  (that is, if all modes of set  $\alpha$  have the same kinetic energy in the ensemble average) and if a similar statement holds about the  $\beta$  modes, then one obtains from Eqs. (78) the simple result

$$\tilde{\phi}_{\alpha\beta} = \tilde{\phi}_{\beta\alpha} = \sum_{j=1}^{N_{\alpha}} \sum_{k=1}^{N_{\beta}} \overline{\phi_{\alpha j, \beta k}} \quad (79)$$

Equation (79), which is analogous to Eq. (69), indicates that the coupling coefficients to be used in Eq. (77) are equal to the sum of the ensemble average (or the ensemble average of the sum) of the mode-to-mode coupling factors. Of course, Eq. (79) applies only if the conditions used in its derivation are met, that is, if the modal kinetic energies and mode-to-mode coupling factors are statistically independent, and if the ensemble average modal kinetic energies are independent of mode number (constant within a type of set).

### Summary and Conclusions

In this section it has been shown that the steady-state time-average power flow from one set of modes to another is proportional (or approximately proportional) to the difference between the set-average modal kinetic energies of the two sets, provided that both

- 1) the mode-to-mode coupling satisfies the conditions discussed under the heading of "Power Flow Between Two Modes", so that the power flows between modes of the two sets may be taken as proportional to the modal kinetic energy differences;

and

- 2) either a) the mode-to-mode coupling is (at least approximately) the same for all mode pairs;  
or b) all modes in a set have (at least approximately) equal time-average kinetic energies.

Then the factor of proportionality; i.e., the coupling factor, applicable to set-to-set power flow is equal to the sum of all the mode-to-mode coupling factors (or to the product of the number of mode pairs and of the average value of all of the individual mode-to-mode coupling factors).

It has also been shown that the ensemble average of the steady-state time average power flows from one set of an ensemble to a set of another ensemble is proportional to the difference between the ensemble average of the set average modal kinetic energies of the two sets, - provided (1) that the first of the previously listed conditions is satisfied, (2) that the mode-to-mode coupling coefficients and modal kinetic energies are statistically independent, and (3) that the modal kinetic energies in the ensemble average are uniformly distributed within a set. The applicable coupling factor then is the ensemble average of the sum (or the sum of the ensemble average) of all the individual mode-to-mode coupling factors.

The mode-to-mode coupling is essentially the same for all mode pairs if all modes of the  $\alpha$  set are similar to each other, if all modes of the  $\beta$  set are similar to each other, and if the same (or similar) physical mechanisms (e.g., the same type of sound-to-structure coupling) couples all modes of the  $\alpha$  set to all modes of the  $\beta$  set.

All modes in a set have equal kinetic energies (i.e., "equipartition of energy" occurs among the modes of a set), if

- either a) the modes of the set are uniformly excited and either weakly or uniformly coupled to modes of other sets;
- or b) the modes are lightly damped, not excited directly, and uniformly coupled to a set of modes within which all modes have nearly the same energy.

In the special cases discussed in this section, and only in these cases, the average response and dynamic interaction of mode sets can be reduced to simple terms, represented schematically in Fig. 4. The most important expressions pertaining to these special cases are also summarized in this figure. In analyzing the total interaction of two systems by means of the statistical energy approach, one must generally classify the modes of the two systems into sets, such that the modes within each set are sufficiently similar so as to satisfy the conditions which permit this simple treatment of set interaction to be applied.



## LOSS FACTORS, RESISTANCES, AND MODAL DENSITIES

The foregoing development of the concepts of the statistical energy method was presented in terms of power flow coupling coefficients and in terms of the number of modes in a set. These parameters are relatively unfamiliar, and generally more difficult to estimate, calculate, or measure than such more commonly used quantities as loss factors, coupling resistances, and modal densities. It is the purpose of the present section to present the relations between these more common parameters and those used in the previous sections, so that one may restate the previously derived results in terms of the more familiar quantities.

The relations between these two sets of parameters are simple, so that restatement of the previous results in terms of the more common quantities involves only trivial algebraic manipulations. Hence, no restated results are given here. Some applications of the relations given in the present section are illustrated in the subsequent one.

### Loss Factors

The loss factor is classically defined on the basis of a single-degree-of-freedom system oscillating sinusoidally at resonance in the steady state. The definition of the loss factor  $\eta$  may be stated as

$$\eta \equiv E_d / 2\pi E \quad , \quad (80)$$

where  $E_d$  denotes the energy dissipated per cycle by the system and  $E$  denotes the "energy of vibration", which is equal essentially to the total (kinetic plus potential) energy at any time (and is nearly constant), to the time-wise maximum kinetic energy, and also to the time-wise maximum potential energy of the system (Refs. 13, 14).

One may express the loss factor in terms of the time-average power  $D$  dissipated by the system, since  $D = E_d \omega_0 / 2\pi$ , where  $\omega_0$  denotes the radian natural frequency of the system so that  $\omega_0 / 2\pi$  represents the natural frequency in cycles/time. One obtains

$$\eta = \frac{D}{\omega_0 E} = \frac{D}{2\omega_0 T} \quad (81)$$

where  $T$  denotes the time-average kinetic energy and  $T=E/2$  holds for the postulated oscillations. Since for a simple mass-spring-dashpot system  $D = c\langle v^2 \rangle$  and  $T = \langle v^2 \rangle m/2$ , one may readily verify that for such a system

$$\eta = \frac{c}{m\omega_0} = \frac{c}{\sqrt{km}} = 2 \frac{c}{c_c} \quad , \quad (82)$$

where  $c_c$  denotes the critical viscous damping coefficient of the system. Equation (82) involves the viscous damping coefficient  $c$ , and hence applies only for damping of the viscous type; Eq. (82) thus is less general than Eqs. (80) and (81), which are not restricted to any particular type of energy dissipation mechanism. Since the loss factor applies for all damping mechanisms, it is generally preferable to more restricted measures of damping.

A broadened interpretation of Eq. (81) is often used to define (and measure) loss factors. In this interpretation  $\omega_0$  is taken as the center frequency of a band encompassing the system natural frequency,  $E$  is taken as the system energy in that band, and  $D$  is taken as the power dissipated by or (transferred away from) the system in the same band. This interpretation is generally not needed for dealing with a single-degree-of-freedom system, but is useful when one is concerned with a collection of such systems, - i.e., with a set of modes. For a set of modes,  $D$  of Eq. (81) is taken as the total power dissipated by all modes and  $T$  as total kinetic energy of all modes. For a set  $\alpha$  containing  $N_\alpha$  individual modes, the total power dissipation  $D_{T\alpha}$  and total time-average kinetic energy  $T_{T\alpha}$  are given by

$$D_{T\alpha} = \sum_{j=1}^{N_\alpha} D_{\alpha j} = 2\omega_0 \sum_{j=1}^{N_\alpha} \eta_{\alpha j} T_{\alpha j} \quad , \quad (83)$$

$$T_{T\alpha} = \sum_{j=1}^{N_\alpha} T_{\alpha j} = N_\alpha T_\alpha$$



where  $\eta_{\alpha j}$  denotes the loss factor of the  $\alpha j$  mode, as defined by Eq. (81), and  $T_{\alpha}$  denotes the set-average kinetic energy, as in Eq. (64). The loss factor  $\eta_{\alpha}$  of set  $\alpha$  then obeys

$$\eta_{\alpha} \equiv \frac{D_{T\alpha}}{2\omega_0 T_{T\alpha}} = \frac{D_{T\alpha}}{2\omega_0 N_{\alpha} T_{\alpha}} \quad (84)$$

$$= \left( \sum_{j=1}^{N_{\alpha}} \eta_{\alpha j} T_{\alpha j} \right) \div \left( \sum_{j=1}^{N_{\alpha}} T_{\alpha j} \right) ,$$

where now  $\omega_0$  is defined as the center frequency of a band which encompasses all modal resonances. The last form of the above equation indicates that the set loss factor  $\eta_{\alpha}$  is a weighted average of the individual modal loss factor  $\eta_{\alpha j}$ , with the modal kinetic energies  $T_{\alpha j}$  acting as weighting factors.

### Coupling Loss Factors

Mechanical energy may be lost from a given system not only by being dissipated (i.e., being changed to thermal energy), but also by being transferred to another system. Energy conducted from a system is lost to it, just like energy dissipated by it. Hence, it is useful to define a loss factor for energy conduction analogous to that for energy dissipation.

It is convenient to define a coupling loss factor  $\eta_{\alpha j, \beta k}$  of mode  $\alpha j$  to describe the time-average power flow  $P_{\alpha j, \beta k}$  from mode  $\alpha j$  to mode  $\beta k$  in analogy to Eq. (81) by

$$\eta_{\alpha j, \beta k} \equiv \frac{P_{\alpha j, \beta k}}{2 \omega_0 T_{\alpha j}} \bigg|_{T_{\beta k} = 0} \quad (85)$$

The condition  $T_{\beta k} = 0$  is imposed in order to make this loss factor independent of the modal kinetic energies. In view of Eq. (62), one finds that the coupling loss factor is related to the mode-to-mode coupling factor  $\phi_{\alpha j, \beta k}$  as

$$\eta_{\alpha j, \beta k} = \phi_{\alpha j, \beta k} / 2 \omega_0 \quad (86)$$

Eq. (86) indicates that the same coupling loss factor applies for power flow in either direction (that is,  $\eta_{\alpha j, \beta k} = \eta_{\beta k, \alpha j}$ ) for those various previously discussed cases in which the same coupling factor applies for power flow in either direction ( $\phi_{\alpha j, \beta k} = \phi_{\beta k, \alpha j}$ ).

One may also define an analogous loss factor to account for power flow from one set of modes ( $\alpha$ ) to another ( $\beta$ ). The coupling loss factor  $\eta_{\alpha\beta}$  of set  $\alpha$  is defined as

$$\eta_{\alpha\beta} \equiv \left. \frac{P_{\alpha\beta}}{2\omega_o T_{T\alpha}} \right|_{T_{\beta}=0}, \quad (87)$$

where  $\omega_o$  denotes the center frequency of a band which encompasses all modal resonances of both sets. In view of Eqs. (61) and (83), one may rewrite the foregoing as

$$\eta_{\alpha\beta} = \frac{\phi_{\alpha\beta} T_{\alpha}}{2\omega_o N_{\alpha} T_{\alpha}} = \frac{\phi_{\alpha\beta}}{2\omega_o N_{\alpha}}. \quad (88)$$

Here it is important to note that because of the appearance of  $N_{\alpha}$  in Eq. (88) the coupling loss factor  $\eta_{\beta\alpha}$  of set  $\beta$  (for power flow from  $\beta$  to  $\alpha$ ) is generally not the same as the loss factor  $\eta_{\alpha\beta}$  of set  $\alpha$  (for power flow from  $\alpha$  to  $\beta$ ), even if the coupling coefficients  $\phi_{\alpha\beta} = \phi_{\beta\alpha}$  are the same for both flow directions (as in the previously analyzed cases). Instead, in view of Eq. (88) and an expression analogous to it,

$$\eta_{\alpha\beta} N_{\alpha} = \eta_{\beta\alpha} N_{\beta} = \phi_{\alpha\beta} / 2\omega_o. \quad (89)$$

### Loss and Coupling Resistances

In analogy to electrical circuits, in which the time-average power dissipation is given by  $\langle I^2 \rangle R_e$ , where  $I$  denotes current and  $R_e$  electrical resistance, one may define a mechanical or acoustic resistance  $R$  of an elastic system so that the power loss  $D$  from the system is given by

$$D = R \langle V^2 \rangle. \quad (90)$$



For a viscously damped single-degree-of-freedom system the resistance is equal to the viscous damping coefficient  $c$ , of course. In view of Eq. (82) one may relate this resistance to the loss factor  $\eta$  and obtain

$$R = c = \eta m \omega_0 \quad (91)$$

for a system of mass  $m$  and natural frequency  $\omega_0$ .

One may define a coupling resistance  $R_{\alpha j, \beta k}$  which describes the power loss of a mode  $\alpha j$  to a mode  $\beta k$  analogously to Eq. (90) by

$$R_{\alpha j, \beta k} \langle v_{\alpha j}^2 \rangle = P_{\alpha j, \beta k} \Big|_{T_{\beta k} = \langle v_{\beta k}^2 \rangle = 0} \quad (92)$$

In view of Eqs. (85) and (30) one may determine that the coupling resistance is related to the corresponding loss factor as

$$R_{\alpha j, \beta j} = \eta_{\alpha j, \beta j} m_{\alpha j} \omega_0 \quad (93)$$

Since this expression is seen to involve the modal mass  $m_{\alpha j}$ , one observes that in general  $R_{\alpha j, \beta k} \neq R_{\beta k, \alpha j}$ , although the corresponding two loss factors may be equal.

The loss resistance  $R_\alpha$  applicable to a set  $\alpha$  of modes should describe the total power loss according to an equation like Eq. (90). The choice of what one defines as the mean square velocity is somewhat arbitrary; it turns out that in dealing with a set of modes of an elastic system of mass  $m_\alpha$  it is convenient to choose  $\langle v_\alpha^2 \rangle = 2 T_{T\alpha} / m_\alpha$ . Then one obtains

$$R_\alpha = \frac{D_{T\alpha} m_\alpha}{2 T_{T\alpha}} = \eta_\alpha m_\alpha \omega_0 \quad , \quad (94)$$

in view of Eq. (84). The above expression is analogous to Eqs. (91) and (93), except that  $m_\alpha$  here represents a system mass, rather than a modal mass.<sup>9</sup>

<sup>9</sup>With the mode shape normalization indicated in Eq. (7), the modal mass is the same as the system mass. But other normalizations may be used legitimately, and with those this equality would not hold.

By analogy to Eq. (94) one may define a coupling resistance  $R_{\alpha\beta}$  to account for the loss of power from a set of modes  $\alpha$  to a set of modes  $\beta$  as

$$R_{\alpha\beta} \equiv \left. \frac{P_{\alpha\beta} m_{\alpha}}{2 T_{T\alpha}} \right|_{T_{\beta}=0} \quad (95)$$

and determine by use of Eqs. (87) and (88) that

$$R_{\alpha\beta} = \eta_{\alpha\beta} m_{\alpha} \omega_0 = \frac{\phi_{\alpha\beta} m_{\alpha}}{2 N_{\alpha}} \quad (96)$$

Again,  $R_{\alpha\beta}$  and  $R_{\beta\alpha}$  are generally different, even if  $\phi_{\alpha\beta} = \phi_{\beta\alpha}$ .

### Modal Densities

One usually studies the interactions of elastic systems on a band-by-band basis, by determining the power flow from modes of one system which have their resonances in a given frequency band to a set of modes of another system which have their resonances in the same band. One may in general choose any convenient bandwidth, and one may often wish to use different bandwidths for different calculations. The number of modes which fall within the various sets depends on the bandwidths one chooses, so that it is useful to introduce the concept of modal density, in terms of which the dependence of some of the previous results on bandwidth can be stated more explicitly.

In addition, it turns out that modal densities for many systems can be estimated readily; thus, use of modal densities increases the practical utility of the analysis method discussed in this report.

The modal density of an elastic system is defined as the average number of modes per unit frequency interval. If a system exhibits  $N_{\alpha}$  modes whose resonances fall within a frequency interval  $\Delta\omega$ , then the modal density of the system at the center frequency  $\omega$  of the interval is defined as

$$n_{\alpha}(\omega) = N_{\alpha}/\Delta\omega \quad (97)$$



For most purposes it suffices to consider  $n_\alpha(\omega)$  as given by Eq. (97) to be a "smooth" function of frequency. Such smoothness may be attained by restricting the applicability of Eq. (97) to cases where the frequency interval of interest encompasses several modes.<sup>10</sup>

The modal densities of systems may be determined from their "frequency equations", i.e., from the equations which give the system resonance frequencies as a function of the system parameters. For simple systems these calculations may be carried out without great difficulty, as illustrated in Ref. 5.

Table I lists the modal densities of some uniform elastic systems, as obtained from Ref. 5 and from calculations based on Ref. 2. The expressions listed in the table apply strictly only for frequencies which are considerably above the system fundamental (by perhaps at least two octaves), where boundary conditions have no important effects; however, these expressions generally also provide reasonable estimates for the modal densities at frequencies only slightly above the fundamental.

The modal density of a composite system is approximately equal to the sum of the modal densities of the component systems. Thus, for example, the modal density of a plate with attached beams is roughly equal to the modal density of the plate by itself, plus the modal densities of all of the beams by themselves. This additive property of modal densities has not been validated analytically, but appears plausible on the basis of some limited experimental evidence and the following reasoning (Ref. 17).

---

<sup>10</sup>Otherwise, small changes in  $\omega$  might result in large stepwise changes in  $n_\alpha$ . For example, if  $\Delta\omega$  is small, it might encompass one or more resonances for one center frequency, but none for another frequency near the previous one. The assumption of smoothness generally introduces no important errors in estimates of average responses in broad frequency bands. (The errors due to this assumption usually are much smaller than those introduced by other approximations and estimates one usually must make in order to apply the statistical energy approach to practical problems.) However, one may need to concern oneself with the fluctuations in  $n_\alpha$  in dealing with more advanced problems concerning the deviations of response from the broad-band average.

TABLE I

## MODAL DENSITIES OF SOME UNIFORM SYSTEMS\*

| <u>System</u>                   | <u>Motion</u>          | <u>Modal Density, <math>n(\omega)</math></u>  | <u>Auxiliary Expressions</u>  |
|---------------------------------|------------------------|---|---|
| String                          | Lateral                | $L/\pi c_s$   | $c_s = \sqrt{T/\rho A}$   |
| Shaft, Beam                     | Torsion                | $L/\pi c_T$   | $c_T = \sqrt{Gk/\rho J}$  |
| Shaft, Beam                     | Longitudinal           | $L/\pi c_\ell$  | $c_\ell = \sqrt{E/\rho}$  |
| Beam                            | Flexure                | $\frac{L}{2\pi} (\omega \kappa_b c_\ell)^{-1/2}$  | $\kappa_b c_\ell = \sqrt{EI/\rho A}$  |
| Membrane                        | Lateral                | $A_s \omega / 2\pi c_m^2$   | $c_m = \sqrt{S/ph}$   |
| Plate                           | Flexure                | $A_s / 4\pi \kappa_p c_\ell$  | $\kappa_p c_\ell = \sqrt{D/\rho h} = \sqrt{Eh^2/12\rho(1-\nu^2)}$                           |
| Room,<br>(Acoustic Volume)      | Sound<br>(Compression) | $V_o \omega^2 / 2\pi^2 c_a^3$   |   |
| Cylindrical Shells<br>(Ref. 15) | Flexure                | $\begin{cases} \approx n_p & \text{for } \omega/\omega_r > 1 \\ \approx n_p \left(\frac{\omega}{\omega_r}\right)^{2/3} & \text{for } \omega/\omega_r < 1 \end{cases}$ | $\begin{aligned} \omega_r &= c_\ell / a \\ n_p &= A_s / 4\pi \kappa_p c_\ell \end{aligned}$ |
| Doubly Curved<br>Shells         | Flexure                | Expressions are complex,<br>given in Ref. 16.   |   |

---

\*See next page for definitions of symbols.



### Symbol Definitions for Table I

|            |   |
|------------|---|
| $A$        | cross-section area                        |
| $A_s$      | surface area                              |
| $c_a$      | acoustic wave velocity                    |
| $c_l$      | longitudinal wave velocity                |
| $c_m$      | membrane wave velocity                    |
| $c_s$      | string wave velocity                      |
| $c_T$      | torsional wave velocity                   |
| $D$        | plate rigidity                            |
| $E$        | Young's modulus                           |
| $G$        | shear modulus                             |
| $h$        | thickness                                 |
| $I$        | centroidal moment of inertia of $A$       |
| $J$        | polar moment of inertia of $A$            |
| $K$        | torsional constant of $A$                 |
| $L$        | length                                    |
| $S$        | membrane tension force/unit edge length   |
| $T$        | string tension force                      |
| $V_o$      | volume                                    |
| $\kappa_b$ | radius of gyration of $A$                 |
| $\kappa_p$ | radius of gyration of plate cross section |
| $\nu$      | Poisson's ratio                           |
| $\omega$   | frequency (radians/time)                  |
| $\rho$     | material density                          |

The number of modes in a lumped parameter system (e.g., one consisting of ideal springs, masses, and dashpots) is equal to the number of degrees of freedom. If one interconnects two such systems, then the total number of degrees of freedom of the composite system is equal (or very nearly equal, depending on the manner of interconnection) to the sum of the number of degrees of freedom of the individual systems. Thus, the number of modes of the combined system is equal to the sum of the numbers of modes of the two constituent systems.

However, some additional considerations are required, since the modal resonances of the combined system generally occur at different frequencies than the resonances of the individual systems. If one interconnects two single-degree-of-freedom systems having different resonant frequencies, then one obtains a two-degree-of-freedom system which has one resonance below the lower of the two individual system resonances, and one which is above the higher of the individual resonances. In other words, the interconnection in effect shifts the resonances both upward and downward. If two multimodal systems are interconnected, it is likely that on the average as many resonances are shifted into a fixed frequency band as are shifted out of it, so that the total number of modes in the band is unchanged by the interconnection process.



## SOME SPECIAL RESULTS FOR TWO COUPLED MODE SETS

For two coupled mode sets, where one of the sets is not directly excited, use of the statistical energy method is particularly fruitful and can lead to valuable qualitative insights, as well as to quantitative results. The present section summarizes general results applicable to the aforementioned case, points out some conclusions one may draw from them, and illustrates application of these results to some special cases.

### Vibrations of Indirectly Excited Mode Sets

Consider two coupled sets of modes, as represented schematically in Fig. 4, and assume that  $A_\beta=0$ , - that is, that no energy is supplied to mode set  $\beta$  except via its coupling to mode set  $\alpha$ . In the steady state the total power  $D_{T\beta}$  dissipated by set  $\beta$  must be equal to the power  $P_{\alpha\beta}$  supplied to it from set  $\alpha$ . If the modes of each set are sufficiently "similar", in the sense indicated in the previous chapter, so that one may validly use Eq. (61) with  $\phi_{\alpha\beta}=\phi_{\beta\alpha}$  to describe  $P_{\alpha\beta}$ , then the energy balance for set  $\beta$  requires that

$$\phi_{\alpha\beta}(T_\alpha - T_\beta) = 2 \omega_o \eta_\beta N_\beta T_\beta \quad , \quad (98)$$

in view of Eq. (84). If one solves this expression for  $T_\beta$  one obtains a relation between the average modal kinetic energies of the two sets which may be expressed as

$$\boxed{\frac{T_\beta}{T_\alpha} = \frac{(\phi_{\alpha\beta}/2\omega_o)}{(\phi_{\alpha\beta}/2\omega_o) + \eta_\beta N_\beta} = \frac{\eta_{\beta\alpha}}{\eta_{\beta\alpha} + \eta_\beta}} \quad (99)$$

where Eq. (89) has been used to replace  $\phi_{\alpha\beta}$  by the coupling loss factor<sup>11</sup>  $\eta_{\beta\alpha}$ .

<sup>11</sup>It should be noted that Eq. (99) involves  $\eta_{\beta\alpha}$ , the loss factor which pertains to energy flow from  $\beta$  to  $\alpha$ , whereas the actual energy flow must be in the other direction if set  $\beta$  is not directly excited.

As indicated in the discussion pertaining to Eqs. (17) and (18), the space- and time-average mean square velocity  $\langle \bar{v}_\alpha^2 \rangle$  associated with  $N_\alpha$  modes of an elastic system of mass  $m_\alpha$  is related to the total kinetic energy  $T_{T\alpha}$  of these modes as

$$T_{T\alpha} = N_\alpha T_\alpha = \frac{1}{2} m_\alpha \langle \bar{v}_\alpha^2 \rangle \quad . \quad (100)$$

If one uses this expression and an analogous one for set  $\beta$  to eliminate the average modal energies  $T_\alpha$  and  $T_\beta$  from Eq. (99), one obtains the result

$$\frac{\langle \bar{v}_\beta^2 \rangle}{\langle \bar{v}_\alpha^2 \rangle} = \frac{N_\beta}{N_\alpha} \cdot \frac{m_\alpha}{m_\beta} \cdot \frac{\eta_{\beta\alpha}}{\eta_{\beta\alpha} + \eta_\beta} = \frac{n_\beta(\omega)}{n_\alpha(\omega)} \cdot \frac{m_\alpha}{m_\beta} \cdot \frac{\eta_{\beta\alpha}}{\eta_{\beta\alpha} + \eta_\beta} \quad . \quad (101)$$

The last form of this expression is obtained by use of Eq. (97) and is based on the assumption that the  $N_\alpha$  and  $N_\beta$  modes correspond to the same frequency band  $\Delta\omega$ .

Equation (101) is a surprisingly simple and useful result, which permits one to calculate the mean square velocity of the indirectly excited elastic system from that of the directly excited one if one knows the masses of the two systems, their modal densities (e.g., from Table I) in the various frequency bands of interest, and the coupling and dissipation loss factors. A further simplification occurs for those cases where  $\eta_\beta \ll \eta_{\beta\alpha}$ , since then Eq. (101) reduces to a relation which does not involve the loss factors. Applications of Eq. (101) to a beam coupled to a plate and to two coupled plates, and related experimental results showing generally good agreement with theoretical predictions, are presented in Ref. 18.

The grouping of the loss factor terms appearing in Eqs. (99) and (101) permits one to make some observations concerning the effectiveness of adding damping to indirectly excited systems. If the damping of the  $\beta$  system is small initially, that is, if  $\eta_\beta \ll \eta_{\beta\alpha}$ , then added damping will reduce the system vibrations

$\langle \bar{v}_\beta^2 \rangle$  only if this added damping (or increase in  $\eta_\beta$ ) is significant as compared to  $\eta_{\beta\alpha}$ . This observation may explain why the addition of some damping to a structure (system  $\beta$ ) excited by sound in an acoustic space (system  $\alpha$ ) may have little effect on some components of the structural vibrations.



On the other hand, Eq. (101) indicates that the mean square velocity of the indirectly excited system varies approximately inversely with the loss factor  $\eta_\beta$  in those cases where  $\eta_\beta \gg \eta_{\beta\alpha}$ . Thus, added damping is effective in reducing vibrations of indirectly excited systems, provided that the resulting dissipation loss factor exceeds the coupling loss factor.

From Eq. (99) or (98) one may conclude that  $T_\beta \approx T_\alpha$  if  $\eta_{\beta\alpha} \gg \eta_\beta$ . That is, if the power dissipation of set  $\beta$  is negligible, then so is the power flow from set  $\alpha$  to set  $\beta$ , and the average kinetic energy  $T_\beta$  of the indirectly excited set will approach that of the directly excited set. Thus, if  $\eta_{\beta\alpha} \gg \eta_\beta$ , then the two mode sets will approach "modal energy equilibrium" or "modal energy equipartition".

### Effect of Coupling on Vibrations of Directly Excited Mode Sets

The present discussion, like that of the previous section, deals with two coupled sets of modes, whose interaction may be visualized by means of the lower portion of Fig. 3. Here, as in the previous section, no energy is assumed to be supplied to set  $\beta$  from outside the system; i.e.  $A_\beta = 0$ . However, unlike the previous section, which focused on the indirectly excited set  $\beta$ , the present section is concerned with the directly excited set  $\alpha$ .

Conservation of energy requires that the time-average power  $A_\alpha$  supplied to set  $\alpha$  in the steady state be equal to the total power dissipated by both mode sets. As previously pointed out, the power dissipated by set  $\beta$  must be equal to the power supplied to it from set  $\alpha$ . Thus, one may write

$$\begin{aligned} A_\alpha &= 2\omega_0 \eta_\alpha N_\alpha T_\alpha + \phi_{\alpha\beta} (T_\alpha - T_\beta) \\ &= 2\omega_0 T_\alpha \left[ \eta_\alpha N_\alpha + \frac{\phi_{\alpha\beta}}{2\omega_0} \left( 1 - \frac{T_\beta}{T_\alpha} \right) \right] \end{aligned} \quad (102)$$

If one eliminates  $T_\beta$  by substituting Eq. (99), and if one introduces the coupling loss factors by use of Eq. (89), then one may find that the foregoing results in

$$\frac{A_\alpha}{2\omega_o T_\alpha N_\alpha} \equiv \eta_{\alpha \text{ app}} = \eta_\alpha + \frac{\eta_{\alpha\beta} \eta_\beta}{\eta_{\beta\alpha} + \eta_\beta} \quad (103)$$

The "apparent loss factor"  $\eta_{\alpha \text{ app}}$  introduced in the foregoing equation corresponds to the value of the dissipation loss factor which one would ascribe to set  $\alpha$  (on the basis of measurements performed on set  $\alpha$ ) if one were not aware that this set is coupled to set  $\beta$ . In order to determine the loss factor  $\eta_\alpha$  of set  $\alpha$  in absence of coupling, one would turn to Eq. (84) and determine the values of the various parameters  $N_\alpha$ ,  $T_\alpha$ ,  $D_{T\alpha}$  that appear there. Since the directly supplied average power  $A_\alpha$  must be equal to the total dissipated  $D_{T\alpha}$  (in absence of coupling to other mode sets), one may replace  $D_{T\alpha}$  by  $A_\alpha$ ; then the ratio appearing in Eq. (84) becomes precisely that which occurs on the left-hand side of Eq. (103).

It is evident from Eq. (103) that the apparent loss factor  $\eta_{\alpha \text{ app}}$  is never smaller than the actual dissipation loss factor  $\eta_\alpha$ .

Thus, if one is unaware that a system whose loss factor one is measuring is coupled to another system, then one always obtains a loss factor value which is too large. The error is insignificant, however, if the coupling is poor ( $\eta_{\alpha\beta}, \eta_{\beta\alpha} \approx 0$ ) and/or if the coupled set is relatively lossless ( $\eta_\beta \ll \eta_\alpha$ ). Effects of energy conduction on loss-factor measurements are discussed in some more detail in Refs. 19 and 20.

From Eq. (103) one may also deduce that the response  $\langle \bar{V}_\alpha^2 \rangle = 2N_\alpha T_\alpha / m_\alpha$  to a given excitation  $A_\alpha$  is not controlled by the actual dissipation loss factor  $\eta_\alpha$  (as in the uncoupled case), but rather by the apparent loss factor  $\eta_{\alpha \text{ app}}$ . Thus, the coupled indirectly excited system  $\beta$  increases the effective damping of the directly excited system  $\alpha$ ; - a conclusion which is intuitively "obvious", since system  $\beta$  here serves to extract energy from system  $\alpha$ .

### Interaction of Sound and Structures

Some of the previous results of this section are particularly useful for estimating the sound produced in an acoustic volume due to the vibration of a structure, or for predicting the structural vibrations induced by a diffuse sound field.



Consider the case where a structure S is directly excited by random forces acting on it, and where these structural vibrations produce noise in a room R in which the structure is located. Here the structural modes in a frequency band of interest correspond to a directly excited set  $\alpha$ , the modes of the acoustic space R correspond to the indirectly excited set  $\beta$ .

The total energy  $E_R$  in a diffuse acoustic field within fluid volume  $V_0$  in a frequency band  $\Delta\omega$  is related to the mean square acoustic pressure  $\langle p^2 \rangle$  measured in the same frequency band as (p. 95 of Ref. 5)

$$E_R = \langle p^2 \rangle V_0 / \rho_a c_a^2 \quad (104)$$

where  $\rho_a$  denotes the density and  $c_a$  the sound velocity of the fluid. This total energy is on the average half potential and half kinetic, hence the average total kinetic energy  $T_{TR}$  is one half of the total. The average modal kinetic energy  $T_R$  may be obtained by dividing  $T_{TR}$  by  $n_R \cdot \Delta\omega$ , the number of modes in the interval. With the modal density expression given in Table I (p. 48) for acoustic volumes, one thus finds the average kinetic energy of an acoustic mode of a room to be given by

$$T_R = \frac{E_R}{2 n_R \Delta\omega} = \frac{\pi^2 c_a}{\rho_a \omega^2} \frac{\langle p^2 \rangle}{\Delta\omega} \quad (105)$$

If one applies Eq. (100) to the structure under consideration and replaces the number of modes  $N_S$  by  $n_s \cdot \Delta\omega$  according to Eq. (97), then one finds that the average kinetic energy of a structural mode is given by

$$T_S = \frac{m_s \langle \bar{v}_s^2 \rangle}{2 n_s \cdot \Delta\omega} = \frac{m_s \langle \bar{a}_s^2 \rangle}{2 \omega^2 n_s \Delta\omega} \quad (106)$$

where  $\langle \bar{v}_s^2 \rangle$  represents the mean square velocity and  $\langle \bar{a}_s^2 \rangle = \omega^2 \langle \bar{v}_s^2 \rangle$  the mean square acceleration of the structure measured in the frequency band  $\Delta\omega$  with center frequency  $\omega$ .

Introduction of Eqs. (105) and (106) into Eq. (99), with R taking the place of the indirectly excited system  $\beta$  and S that of the directly excited system  $\alpha$ , leads to the following relation between the mean square values of the acoustic pressure and the structural acceleration:

$$\frac{\langle \bar{p}^2 \rangle}{\langle \bar{a}_s^2 \rangle} = \frac{\rho_a m_s}{2\pi^2 c_a n_s} \frac{\eta_{RS}}{\eta_{RS} + \eta_R} \quad (107)$$

In most cases the room-to-structure coupling loss factor  $\eta_{RS}$  may be expected to be much smaller than the dissipation loss factor of the room. With this inequality, and with

$$\eta_{RS} = \frac{\eta_{SR} n_s}{n_R} = \frac{n_s R_{SR}}{n_R m_s \omega} \quad (108)$$

obtained from Eqs. (89), (97), (96), one finds that one may rewrite Eq. (107) as

$$\frac{\langle \bar{p}^2 \rangle}{\langle \bar{a}_s^2 \rangle} = \frac{\rho_a R_{SR}}{2\pi^2 c_a \omega n_R \eta_R} = \frac{R_{SR}}{\eta_R} \cdot \frac{\rho_a c_a^2}{V_o \omega^3} \quad , \quad (109)$$

where the last form is obtained by substituting for  $n_R$  the appropriate expression from Table I. The structure-to-room coupling resistance  $R_{SR}$  corresponds to the "radiation resistance" which acts on the structure; i.e. to the resistance which governs the loss of energy by the structure due to acoustic radiation from it.

One may also consider a situation in which an acoustic volume R is directly excited by a noise source, and where the resulting acoustic pressures cause an (indirectly excited) structure to vibrate. Here the volume excites the structure; i.e. the roles of the two systems here are interchanged from the previously discussed case. Equations (104)-(106) still apply, but now the structure S corresponds to the indirectly excited set  $\beta$  and the room R to the indirectly excited set  $\alpha$ . Here, use of Eq. (99) yields a result which one may write

$$\frac{\langle \bar{a}_s^2 \rangle}{\langle \bar{p}^2 \rangle} = 2\pi^2 \frac{c_a}{\rho_a} \frac{n_s}{m_s} \mu \quad (110)$$

where



$$\mu \equiv \frac{\eta_{SR}}{\eta_{SR} + \eta_s} = \frac{R_{SR}}{R_{SR} + R_s} \quad (111)$$

and the last form of Eq. (111) follows from Eq. (96).

The results given in Eqs. (109)-(111) first appeared in Ref. 10. This reference also describes some experimental results, which are in good agreement with theoretical predictions.

A word of caution is required, however, concerning the applicability of the results of this section. These results are derived from Eqs. (98) and (99), which in turn are based on the assumption that a simple power flow relation holds for the two mode sets considered. Thus, Eqs. (109)-(111) should only be applied to cases where all structural modes in the frequency band of interest are uniformly coupled to all room modes in the band. If differently excited or differently coupled modes occur in the band, one must divide them into groups of like modes, and apply the foregoing results to each group separately, of course using properly modified modal density expressions, as needed. Differences in sound-structure coupling and calculations taking these into account are discussed in Ref. 5.

## CONCLUDING REMARKS

It is hoped that this report will introduce the reader to the basic concepts involved in the statistical energy analysis of vibrating systems, so that he may put this approach to use with some confidence. It was endeavored to describe the most important aspects of this analysis approach in considerable detail, in order to give the reader an understanding of its range of validity. Of course, not all aspects of available results pertinent to this approach could be covered within the time and space allotted to this report; the reader interested in particular topics may do well to consult some of the references given here, as well as some of the background information cited in these references.

Although the results presented in this report are limited to linear coupling, a study of nonlinearly coupled modes has been published recently (Ref. 21). This study shows that the net energy flow between nonlinearly coupled modes also occurs in the direction of the energy gradient, and that no energy flows if the modal energies are equal.

The present report has dealt essentially only with two coupled modes, or with two coupled mode sets, but not with more sets or modes in cascade. For loose coupling, extension of the two-set results to multiset-systems can readily be accomplished. Such extensions and applications of them appear in Refs. 22-24, and a complete electric-circuit analogy for energy flow in multiset-systems is developed in Ref. 25. As yet unpublished calculations indicate that the power flow between any two adjacent modes of three linearly coupled modes in cascade is proportional to the modal energy difference, even if the modes are well coupled; however, here the factors of proportionality generally involve properties of all three modes and thus differ from the two-coupled-mode case.

Calculations on the basis of the statistical energy approach of variances and confidence limits, in addition to the commonly computed averages of the dynamic variables, are presented in Ref. 18. Illustrations of applications of the statistical energy approach and discussions of the problems which have been studied by means of it appear in Refs. 17-19, 22-28.

At present the usefulness of the statistical energy approach in predicting vibration responses under practical circumstances is limited by the amount of information that is available concerning coupling coefficients and loss factors. Approximate determination of sound-to-structure coupling



coefficients (or resistances) often is not very difficult, as indicated in Ref. 5, but calculations of structure-to-structure coupling coefficients may involve considerable effort (e.g., see Ref. 18). A reasonable amount of information on the dissipative loss factors of materials and simple structures is available in the technical literature (e.g., Ref. 13), but the loss factors of built-up structures can still not be estimated with much confidence - although some progress is being made (Ref. 29).

The statistical energy approach is a uniquely simple and powerful tool for providing one with a qualitative understanding of the most important aspects of the vibrations of complex systems. When combined with other, analytically or experimentally derived, information concerning the energy transport and dissipation parameters involved in a given case, this approach also permits one to arrive readily at quantitative answers to complex vibration problems.

## REFERENCES

1. Kin N. Tong, Theory of Mechanical Vibration, John Wiley and Sons, Inc., New York, 1960.
2. E. E. Ungar, "Mechanical Vibrations", Mechanical Design and Systems Handbook, Ed. by H. A. Rothbart. McGraw-Hill Book Company, Inc., New York, 1964.
3. B. Friedman, Principles and Techniques of Applied Mathematics, John Wiley and Sons, Inc., New York, Chapt. 4, 1956.
4. W. R. LePage and S. Seely, General Network Analysis, McGraw-Hill Book Company, Inc., New York, Chapt. 1, 1952.
5. P. W. Smith, Jr., and R. H. Lyon, "Sound and Structural Vibration", NASA CR-160 (March 1965).
6. S. H. Crandall and W. D. Mark, Random Vibration in Mechanical Systems, Academic Press, New York, 1963.
7. T. D. Scharton, "Random Vibration of Coupled Oscillators and Coupled Structures", Doctoral Thesis, Mass. Institute of Technology, October 1965.
8. Lord Rayleigh, Theory of Sound, Dover Publications, New York, Vol. I, p. 104, 1945.
9. P. Morse, Vibration and Sound, McGraw-Hill Book Company, Inc., New York, pp. 12-16, 1948.
10. R. H. Lyon, and G. Maidanik, "Power Flow Between Linearly Coupled Oscillators", J. Acoust. Soc. Am. 34, 623-639 (May 1962).
11. S. H. Crandall, "Statistical Properties of Response to Random Vibration", Ch. 4 of Random Vibration, Ed. by S. H. Crandall, The MIT Press, Cambridge, Mass., pp. 21-32, 1958.
12. D. C. Karnopp, "Basic Theory of Random Vibration", Ch. 1 of Random Vibration, Volume 2, Ed. by S. H. Crandall, The MIT Press, Cambridge, Mass., 1963.



13. B. J. Lazan, "Energy Dissipation Mechanisms in Structures with Particular Reference to Material Damping", Sec. 1 of Structural Damping, Ed. by J. E. Ruzicka, Amer. Soc. Mech. Engrs., New York, 1959.
14. E. E. Ungar and E. M. Kerwin, Jr., "Loss Factors of Viscoelastic Systems in Terms of Energy Concepts", J. Acoust. Soc. Am. 34, 954-957 (July 1962).
15. M. Heckl, "Vibrations of Point-Driven Cylindrical Shells", J. Acoust. Soc. Am. 34, 1553-1557, October 1962.
16. V. V. Bolotin, "On the Density of the Distribution of Natural Frequencies of Thin Elastic Shells", J. Appl. Math. and Mech. 27, No. 2, 538-543 (Transl. from Soviet Journal: Prikladnaya Matematika y Mekhanika, 27, No. 2, pp. 362-364, 1963).
17. G. Maidanik, "Response of Ribbed Panels to Reverberant Acoustic Fields", J. Acoust. Soc. Am. 34, 809-826 (June 1962).
18. R. H. Lyon and E. Eichler, "Random Vibration of Connected Structures", J. Acoust. Soc. Am. 36, 1344-1354, July 1964.
19. E. M. Kerwin, Jr., "Mechanisms and Measurement of Structural Damping", Ship Silencing Symposium, Groton, Connecticut, May 1963.
20. E. E. Ungar "Recommended Studies of Vibratory Energy Dissipation and Conduction in Aerospace Structures", FDL TDR 64-111, Appendix C, (August 1964).
21. D. E. Newland, "Energy Sharing in the Random Vibration of Non-linearly Coupled Modes", J. of Institute of Math. and Its Applications 1, 189-207 (September 1965).
22. E. Eichler, "Thermal Circuit Approach to Vibrations in Coupled Systems and the Noise Reduction of a Rectangular Box", J. Acoust. Soc. Am. 37, 995-1007 (June 1965).
23. R. H. Lyon and T. D. Scharton, "Vibrational-Energy Transmission in a Three-Element Structure", J. Acoust. Soc. Am. 38, 253-261 (August 1965).
24. R. H. Lyon, "An Energy Method for Prediction of Noise and Vibration Transmission", Shock, Vibr. and Assoc. Environments, Bull. No. 33, Pt. II, pp. 13-25 (1964).

25. D. U. Noiseux, R. H. Lyon, C. W. Dietrich, E. A. Starr, "Dynamic Response, Energy Methods, and Test Correlation of Flight Vehicle Equipments", AFFDL-TR 65-92, Vol. I, Appendix III, (May 1965).
26. R. H. Lyon and G. Maidanik, "Statistical Methods in Vibration Analysis", AIAA Journal 2, 1015-1024 (June 1964).
27. P. A. Franken and R. H. Lyon, "Estimation of Sound-Induced Vibrations by Energy Methods, with Applications to the Titan Missile", Shock, Vibration, and Associated Environments, Bull. No. 31, Part III, pp. 12-26, (1963).
28. I. Dyer, "Response of Space Vehicle Structures to Rocket Engine Noise", Ch. 7 of Random Vibration, Vol. 2, Ed. by S. H. Crandall, The MIT Press, Cambridge, Mass., (1963).
29. E. E. Ungar, "Energy Dissipation at Structural Joints; Mechanisms and Magnitudes", FDL-TDR-64-98, (August 1964).



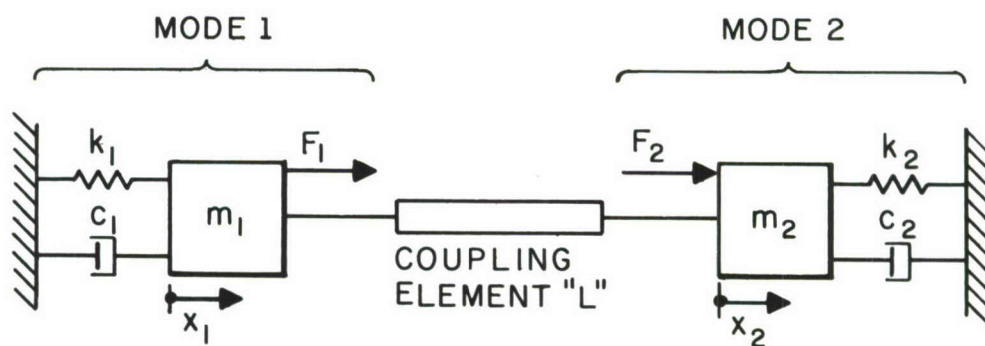


FIG.1 TWO COUPLED MODES

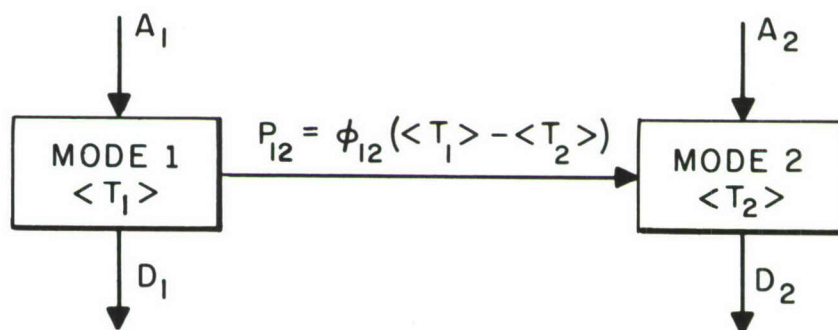


FIG.2 POWER FLOW DIAGRAM FOR TWO COUPLED MODES

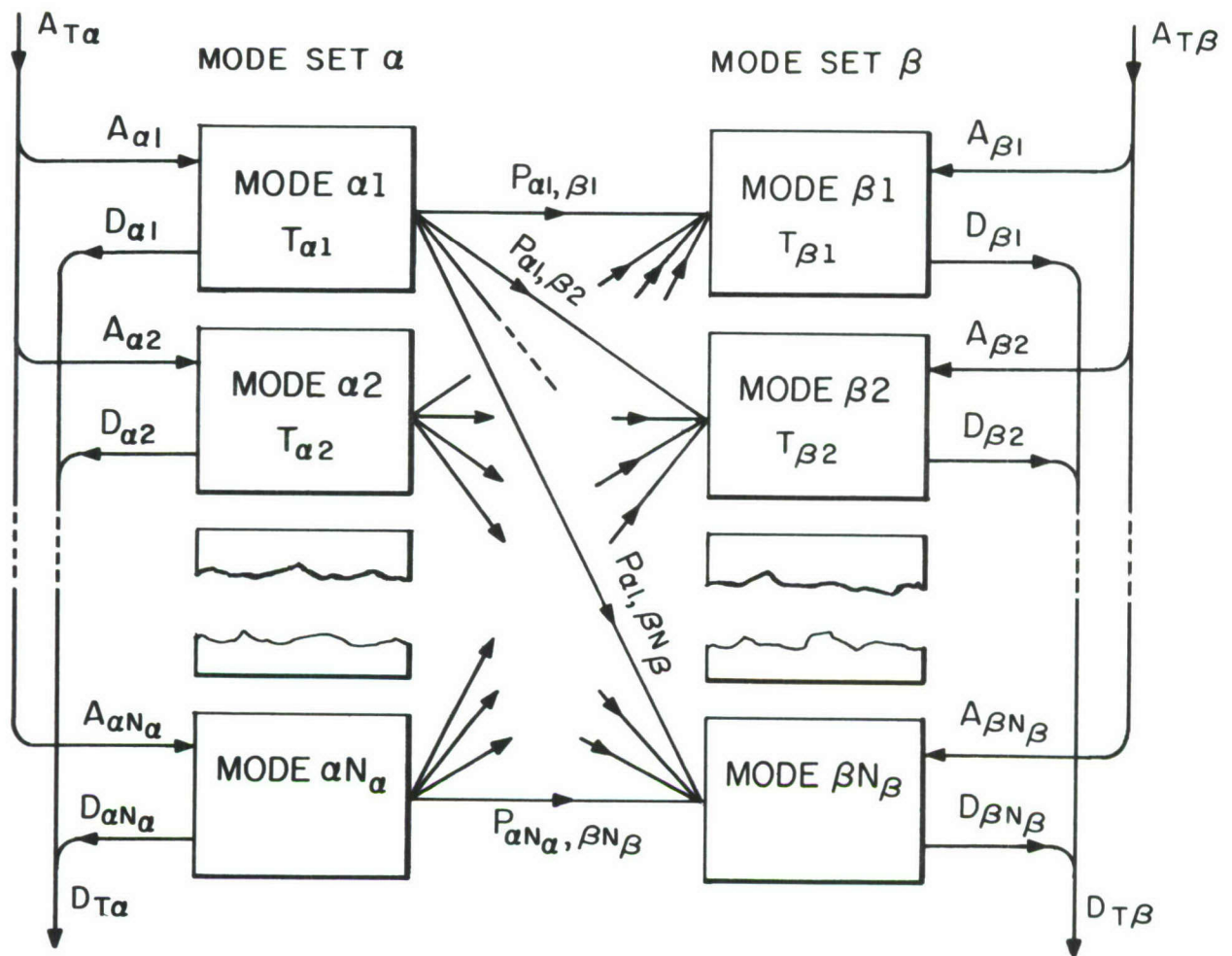
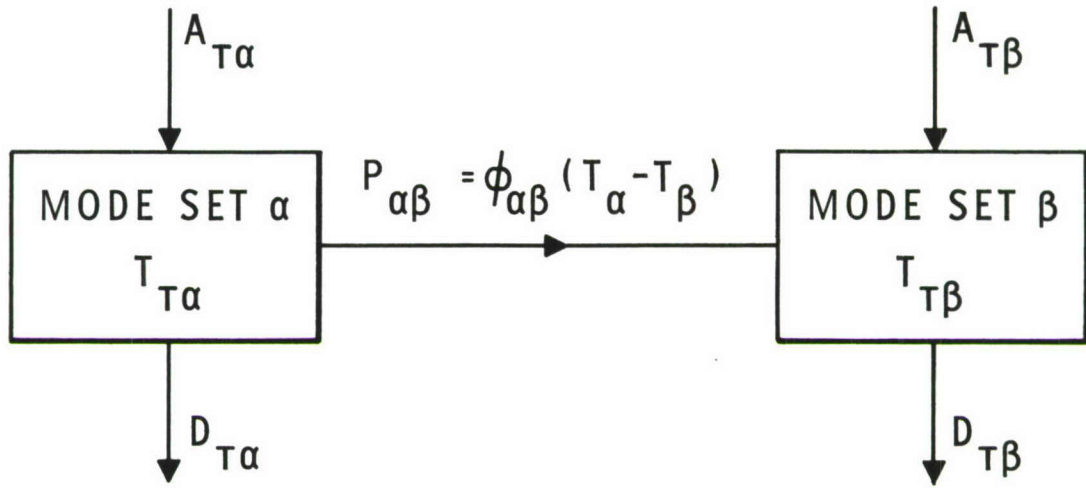


FIG.3 TWO COUPLED SETS OF MODES





$$A_{T\alpha} = N_{\alpha} A_{\alpha} \approx \pi N_{\alpha} S_{\alpha} / m_{\alpha}$$

Ref. Eqs.  
(74)

$$D_{T\alpha} = N_{\alpha} D_{\alpha} = 2\omega_0 \eta_{\alpha} T_{T\alpha}$$

(83), (84)

$$T_{T\alpha} = N_{\alpha} T_{\alpha} = \langle \bar{v}_{\alpha}^2 \rangle m_{\alpha} / 2 = N_{\alpha} \langle v_{\alpha}^2 \rangle m_{\alpha} / 2$$

(17), (18)

$$P_{\alpha\beta} = \phi_{\alpha\beta} (T_{\alpha} - T_{\beta})$$

(61), (68), (69), (79)

$$\phi_{\alpha\beta} = N_{\alpha} N_{\beta} \phi_0 = 2\omega_0 N_{\alpha} \eta_{\alpha\beta} = 2\omega_0 N_{\beta} \eta_{\beta\alpha}$$

(68), (88)

See next page for definitions of symbols.  
See pp. 39, 40 for summary of restrictions  
under which the above relations apply.

FIG. 4 POWER FLOW BETWEEN TWO SETS OF MODES

# SYMBOL DEFINITIONS FOR FIG. 4

The following abbreviations are used in this list of symbols:

|      |              |        |                  |
|------|--------------|--------|------------------|
| t.a. | time-average | tot.   | total            |
| s.a. | set-average  | m.s.α. | mode(s) of set α |

| <u>Symbols</u>                       | <u>Definitions</u>  |
|--------------------------------------|---|
| $A_{T\alpha}$                        | t.a. tot. power supplied to all m.s.α.  |
| $D_{T\alpha}$                        | t.a. tot. power dissipated by all m.s.α.  |
| $T_{T\alpha}$                        | t.a. tot. kinetic energy of all m.s.α.  |
| $A_{\alpha}$                         | s.a. of t.a. power supplied to single m.s.α.  |
| $D_{\alpha}$                         | s.a. of t.a. power dissipated by single m.s.α.  |
| $T_{\alpha}$                         | s.a. of t.a. kinetic energy of single m.s.α.  |
| $N_{\alpha}$                         | number of modes in set α  |
| $m_{\alpha}$                         | tot. mass of elastic system containing the m.s.α.   |
| $\langle \bar{v}_{\alpha}^2 \rangle$ | t.a. and space-average mean-square velocity<br>(of elastic system containing set α) due to all m.s.α. |
| $\langle v_{\alpha}^2 \rangle$       | t.a. and s.a. mean square velocity of single m.s.α.   |
| $S_{\alpha}$                         | s.a. spectral density of forces acting on m.s.α.  |
| $\eta_{\alpha}$                      | dissipation loss factor for set α   |
| $\eta_{\alpha\beta}$                 | loss factor pertaining to power flow from set α to set β  |
| $\omega_0$                           | center frequency of band containing all modal resonances<br>of both sets                              |
| $\phi_{\alpha\beta}$                 | set-to-set power flow coefficient   |
| $\phi_0$                             | mode-to-mode power flow coefficient (averaged over<br>both sets)                                      |



UNCLASSIFIED

Security Classification

## DOCUMENT CONTROL DATA - R&amp;D

(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)

|   |  |  |                       |
|---|--|--|-----------------------|
| 1. ORIGINATING ACTIVITY (Corporate author)<br>Bolt Beranek and Newman Inc.<br>50 Moulton Street<br>Cambridge, Massachusetts 02138   |  | 2a. REPORT SECURITY CLASSIFICATION<br>Unclassified   |                       |
|   |  | 2b. GROUP  |                       |
| 3. REPORT TITLE<br><br>FUNDAMENTALS OF STATISTICAL ENERGY ANALYSIS OF VIBRATING SYSTEMS   |  |  |                       |
| 4. DESCRIPTIVE NOTES (Type of report and inclusive dates)<br>Final Report, July 1965 - February 1966  |  |  |                       |
| 5. AUTHOR(S) (Last name, first name, initial)<br><br>Ungar, Eric E.   |  |  |                       |
| 6. REPORT DATE<br>MAY 1966  |  | 7a. TOTAL NO. OF PAGES<br>66   | 7b. NO. OF REFS<br>29 |
| 8a. CONTRACT OR GRANT NO. AF33(615)-2649  |  | 9a. ORIGINATOR'S REPORT NUMBER(S)<br><br>AFFDL-TR-66-52  |                       |
| b. PROJECT NO. 1370   |  |  |                       |
| c. Task No. 137005  |  | 9b. OTHER REPORT NO(S) (Any other numbers that may be assigned this report)<br>BBN 1350  |                       |
| d.  |  |  |                       |
| 10. AVAILABILITY/LIMITATION NOTICES<br><br>Distribution of this document is unlimited.  |  |  |                       |
| 11. SUPPLEMENTARY NOTES<br><br>Supplement to be issued after experimental program   |  | 12. SPONSORING MILITARY ACTIVITY<br>Air Force Flight Dynamics Laboratory<br>RTD, AFSC<br>Wright-Patterson Air Force Base, Ohio |                       |
| 13. ABSTRACT<br><br>The concepts and relations which form the theoretical foundation of the statistical energy approach to vibration analysis are delineated. The utility of this approach for dealing with complex systems is discussed, and its range of applicability is indicated. Examples are presented which illustrate how the mode-to-mode power flow relations developed in the report permit one to obtain approximate solutions to complex problems simply, on the basis of energy conservation considerations. |  |  |                       |

45433

|  |        |    |        |    |        |    |
|--|--------|----|--------|----|--------|----|
| 14. KEY WORDS  | LINK A |    | LINK B |    | LINK C |    |
|  | ROLE   | WT | ROLE   | WT | ROLE   | WT |
| <p>Vibrations</p> <p>Vibratory Systems</p> <p>Sound Transmission</p> |        |    |        |    |        |    |

#### INSTRUCTIONS

1. **ORIGINATING ACTIVITY:** Enter the name and address of the contractor, subcontractor, grantee, Department of Defense activity or other organization (*corporate author*) issuing the report.

2a. **REPORT SECURITY CLASSIFICATION:** Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Marking is to be in accordance with appropriate security regulations.

2b. **GROUP:** Automatic downgrading is specified in DoD Directive 5200.10 and Armed Forces Industrial Manual. Enter the group number. Also, when applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.

3. **REPORT TITLE:** Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capitals in parenthesis immediately following the title.

4. **DESCRIPTIVE NOTES:** If appropriate, enter the type of report, e.g., interim, progress, summary, annual, or final. Give the inclusive dates when a specific reporting period is covered.

5. **AUTHOR(S):** Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, show rank and branch of service. The name of the principal author is an absolute minimum requirement.

6. **REPORT DATE:** Enter the date of the report as day, month, year; or month, year. If more than one date appears on the report, use date of publication.

7a. **TOTAL NUMBER OF PAGES:** The total page count should follow normal pagination procedures, i.e., enter the number of pages containing information.

7b. **NUMBER OF REFERENCES:** Enter the total number of references cited in the report.

8a. **CONTRACT OR GRANT NUMBER:** If appropriate, enter the applicable number of the contract or grant under which the report was written.

8b, 8c, & 8d. **PROJECT NUMBER:** Enter the appropriate military department identification, such as project number, subproject number, system numbers, task number, etc.

9a. **ORIGINATOR'S REPORT NUMBER(S):** Enter the official report number by which the document will be identified and controlled by the originating activity. This number must be unique to this report.

9b. **OTHER REPORT NUMBER(S):** If the report has been assigned any other report numbers (*either by the originator or by the sponsor*), also enter this number(s).

10. **AVAILABILITY/LIMITATION NOTICES:** Enter any limitations on further dissemination of the report, other than those

imposed by security classification, using standard statements such as:

- (1) "Qualified requesters may obtain copies of this report from DDC."
- (2) "Foreign announcement and dissemination of this report by DDC is not authorized."
- (3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through \_\_\_\_\_."
- (4) "U. S. military agencies may obtain copies of this report directly from DDC. Other qualified users shall request through \_\_\_\_\_."
- (5) "All distribution of this report is controlled. Qualified DDC users shall request through \_\_\_\_\_."

If the report has been furnished to the Office of Technical Services, Department of Commerce, for sale to the public, indicate this fact and enter the price, if known.

11. **SUPPLEMENTARY NOTES:** Use for additional explanatory notes.

12. **SPONSORING MILITARY ACTIVITY:** Enter the name of the departmental project office or laboratory sponsoring (*paying for*) the research and development. Include address.

13. **ABSTRACT:** Enter an abstract giving a brief and factual summary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the information in the paragraph, represented as (TS), (S), (C), or (U).

There is no limitation on the length of the abstract. However, the suggested length is from 150 to 225 words.

14. **KEY WORDS:** Key words are technically meaningful terms or short phrases that characterize a report and may be used as index entries for cataloging the report. Key words must be selected so that no security classification is required. Identifiers, such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical context. The assignment of links, roles, and weights is optional.